Marks: 100

10

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## M.E. MECHANICAL ENGINEERING EXAM 2017

## FIRST YEAR SECOND SEMESTER

## **GAS DYNAMICS**

(Answer any FIVE questions)

Time: Three hours

	Different parts of the same question should be <u>answered together</u> .  All symbols carry their usual meanings unless otherwise mentioned.  Assume any relevant data if necessary.	
1.	a) What do you mean by Mach wave and Mach angle? Explain with suitable diagram.	8
	b) In an isentropic flow of air through duct whose area is decreasing. Determine the	12
	percentage changes in velocity, density and pressure induced by a 2 percentage	
	reduction in area for Mach number 0.2 and 0.8.	
2.	a) A pitot tube is placed in a subsonic air flow. The static pressure and temperature in the	
	flow are 100 kPa and 285 K respectively. Determine the air velocity and the Mach	
	number if the differential mercury manometer shows a reading of 200mm.	10
	b) In one dimensional isentropic flow, air flows through a nozzle of inlet area 10 cm <sup>2</sup> . At	
	the inlet velocity, temperature, pressure are 100 m/s, 30°C and 700kPa. Determine the	
	mass flow rate through the nozzle if exit pressure is 300kPa. Also determine the exit	10
	velocity of the nozzle.	
3.	a) What is normal shock wave? Explain with example.	8
	b) Derive normal shock wave relations in terms of Mach number.	12
4.	a) What is oblique shock wave? Discuss with example.	7
	b) Explain the limiting cases of an oblique shock wave.	6
	c) What do you mean by reflection of oblique shock wave?	7

b) Discuss about reflection and interaction of expansion waves with suitable diagram.

5. a) What do you mean by expansion waves? Explain it.

- 6. a) Explain the flow through a convergent divergent nozzle with under expanded 10 condition.
  - b) Gas flows from a large reservoir through a convergent-divergent nozzle nozzle, the pressure and temperature in the reservoir being respectively 500kPa and 45°C respectively. The throat area of the nozzle is 10<sup>-4</sup> m<sup>2</sup> and exit pressure is 120kPa. Determine the mass flow rate through the isentropic nozzle and the exit area of the 10 nozzle.
- 7. a) Discuss the effect of friction on flow variables in an adiabatic flow in a constant area 10 duct.
  - b) Discuss the effect of heat addition and rejection on flow variables in a constant area duct neglecting viscosity.
- 8. Write short notes on:

4 X 5

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- a) Flow over a bodies involving shock and expansion waves
- b) Fanno Line
- c) Hot wire anemometer
- d) Pitot Tube