

Phasor Diagram of Cylindrical Rotor Synchronous Machine

Two Reaction Theory:


The different values of reluctance along polar axis (d axis) and inter polar axis $n$ ( $q$ axis) , make analysis of salient pole machine different from cylindrical rotor machine. Cylindrical rotor theory is not applicable for salient pole machine as the theory is based on constant air gap reluctance. The effect of salient pole is taken into account in the 'Two-Reaction Theory' proposed by Blondel.

According to two reaction theory the sinusoidal armature reaction m.m.f $F_{a}$ is resolved into two sinusoidal components; $\mathrm{F}_{\mathrm{ad}}$ along the d axis and $\mathrm{F}_{\mathrm{aq}}$ along the q axis. The d axis component $\mathrm{F}_{\mathrm{ad}}$ is magnetizing or demagnetizing depending on the power factor and produces only change in the field strength. Whereas the $q$ axis component $F_{\text {aq }}$ produces only distortion of the field flux wave. $F_{\text {ad }}$ and $F_{\text {aq }}$ are assumed to be produced by components of armature current along the $d$ axis ( $I_{d}$ ) and component along $q$ axis ( $\left(\mathrm{I}_{\mathrm{q}}\right)$ respectively.


Phasor diagram of Salient Pole Synchronous generator based on Two Reaction Theory
Note that as the length of air gap along $d$ axis and $q$ axis are unequal, the reluctances are different, so,

$$
\frac{\emptyset_{a d}}{\emptyset_{a q}} \neq \frac{F_{a d}}{F_{a q}}
$$

$\mathrm{E}_{\text {ad }}$ is induced by $\mathrm{F}_{\text {ad }}$ lagging it by $90^{\circ}$ and $\mathrm{E}_{\text {aq }}$ is induced by $\mathrm{F}_{\text {aq }}$ lagging it by $90^{\circ}$. So,
$\overline{E_{a d}}=-j K_{d} \bar{F} \overline{F_{a d}}=-j C K_{d} \overline{I_{d}}=-j X_{a d} \overline{I_{d}}$ and $\overline{E_{a q}}=-j K_{q} \overline{F_{a q}}=-j C K_{q} \overline{I_{q}}=-j X_{a q} \overline{I_{q}}$
$\mathrm{X}_{\mathrm{ad}}$ and $\mathrm{X}_{\mathrm{aq}}$ are d axis and q axis magnetizing reactance ( armature reaction reactance) of the synchronous machine. $X_{a d}$ takes care of the effect of armature reaction along the $d$ axis and $X_{a q}$ takes care of the effect of armature reaction along the $q$ axis.

The phasor sum of $\mathrm{E}_{\mathrm{f}}, \mathrm{E}_{\mathrm{ad}}$ and $\mathrm{E}_{\mathrm{aq}}$ gives the air gap voltage $\mathrm{E}_{\mathrm{ar}}$.

$$
\overline{E_{a r}}=\overline{E_{f}}+\overline{E_{a d}}+\overline{E_{a q}}
$$

Again $\overline{V_{t}}+\bar{I}_{a} r_{a}+j \bar{I}_{a} x_{a l}=\overline{E_{a r}}$ and $\bar{I}_{d}+\bar{I}_{q}=\bar{I}_{a} \quad \therefore j \bar{I}_{a} x_{a l}=j \bar{I}_{d} x_{a l}+j \bar{I}_{q} x_{a l}$

$$
\begin{gathered}
\therefore \bar{V}_{t}+\overline{I_{a}} r_{a}+j \overline{I_{a}} x_{a l}=\overline{E_{f}}+\overline{E_{a d}}+\overline{E_{a q}} ; \\
\therefore \bar{V}_{t}+\overline{I_{a}} r_{a}+j \bar{I}_{d} x_{a l}+j \overline{I_{q}} x_{a l}=\overline{E_{f}}-j X_{a d} \overline{I_{d}}-j X_{a q} \overline{I_{q}} \\
\therefore \bar{V}_{t}+\overline{I_{a}} r_{a}+j \overline{I_{d}}\left(x_{a l}+X_{a d}\right)+j \overline{I_{q}}\left(x_{a l}+X_{a q}\right)=\overline{E_{f}} \\
\left.\quad \overline{V_{t}}+\overline{I_{a}} r_{a}+j \overline{I_{d}}\left(X_{d}\right)+j \overline{I_{q}( } X_{q}\right)=\overline{E_{f}}
\end{gathered}
$$

Where $X_{d}=x_{a l}+X_{a d}$ is the direct axis synchronous reactance
$X_{q}=X_{a l}+X_{a q}$ is the quadrature axis synchronous reactance


Phasor Diagram of Synchronous Generator
$I_{d}=I_{a} \sin (\delta+\theta)$ and $I_{q}=I_{a} \cos (\delta+\theta)$

However, note that it is not possible to draw the phasor diagram unless the angle between $\mathrm{E}_{\mathrm{f}}$ and $\mathrm{V}_{\mathrm{t}}$ (i.e. angle $\delta$ is known, because to draw the phasor diagram the armature current has to be resolved into two components.

For this purpose, $\mathrm{V}_{\mathrm{t}} ; \mathrm{I}_{\mathrm{a}}$ and $\mathrm{I}_{\mathrm{a}} \mathrm{r}_{\mathrm{a}}$ phasors are drawn and 'ab' is drawn normal to $\mathrm{I}_{\mathrm{a}} \mathrm{r}_{\mathrm{a}}$. As it is $90^{\circ}$ away from $\mathrm{I}_{\mathrm{a}}$, it can be taken equivalent to a reactive voltage drop of $\mathrm{I}_{\mathrm{a}} \mathrm{X}$. Draw 'ac, perpendicular to 'ob'. Oa'b and acb are right angled triangles and therefore, $\angle b a c=\angle \delta+\theta$

$$
\begin{gathered}
\therefore a c=a b \cos (\delta+\theta)=X I_{a} \cos (\delta+\theta)=d e \\
X I_{q}=X_{q} I_{q} \quad \therefore X=X_{q}
\end{gathered}
$$



Thus, $a b=X_{q} I_{a}$. if ob $=E_{f}^{\prime}$ then $\overline{E_{f}{ }^{\prime}}=\bar{V}_{t}+\bar{I}_{a} r_{a}+j \bar{I}_{a} X_{q}$

The phasors $E_{f}^{\prime}$ and $E_{f}$ are are along the same line. So, angle between $E_{f}^{\prime}$ and $V_{t}$ is $\delta$. So, angle $(\delta+\theta)$ is now known and $I_{d} ; I_{q} \quad$ can be drawn.

Now $b d=c d-c b=X_{d} I_{d}-X_{q} I_{a} \sin (\delta+\theta)=X_{d} I_{d}-X_{q} I_{d}=\left(X_{d}-X_{q}\right) I_{d}$

$$
\overline{E_{f}^{\prime}}+\overline{b d}=\overline{E_{f}}=\overline{V_{t}}+\overline{I_{a}} r_{a}+j \overline{I_{a}} X_{q}+j \overline{I_{d}}\left(X_{d}-X_{q}\right)
$$

## Determination of $X_{d}$ and $X_{q}$ :

## Determination of $X_{d}$

For short circuit condition $V_{t}=0$ therefore $I_{q} X_{q}=0 X_{q} \neq 0 \quad \therefore I_{q}=0$ and $I_{d}=I_{s c}$ which leads to $E_{f}=I_{d} X_{d}$.
So $\frac{\text { Open circuit voltage for a field current }}{\text { short circuit current for the same field current }}=X_{d}$

## Slip Test:

The synchronous machine is driven by the prime mover at a speed slightly different from the slip speed. The field winding is kept open and a positive sequence balanced voltage of reduced magnitude (around $25 \%$ of the rated voltage) is applied to the machine. The relative speed between the rotor poles and the stator rotating magnetic field is the difference between the synchronous speed and the rotor speed, i.e. the slip speed. Small low frequency A.C voltages across the open field winding indicate that the field poles and the magnetic field are rotating in the same direction (essential for the test). If the field poles rotate in the opposite direction then negative sequence reactance would be measured.


Phtsical concept of $X_{d}$ and $X_{q}$

The rotor and the rotating magnetic field are rotating at different speed. At one instant when the peak of the armature m.m.f wave is directed along the pole axis ( $d$-axis) the reluctance offered by the small air gap is minimum (refer to the above figure). At this instance the impressed terminal voltage per phase divided by the per phase armature current gives the direct axis synchronous reactance $X_{d}$. After one quarter of the slip cycle the peak of armature m.m.f. wave is coincident with the inter polar axis ( $q$-axis), the reluctance offered by the large air gap is maximum (refer to the above figure). At this instance the impressed terminal voltage per phase divided by the per phase armature current gives the quadrature axis synchronous reactance $\mathrm{X}_{\mathrm{q}}$.


Oscillograms of armature current; armature terminal voltage and e.m.f induced in the open field winding is shown in the above diagram. A much larger slip has been considered for convenience. In practice the slip is very small. When the armature m.m.f. is along the direct axis, the armature flux linkage of the open field winding is maximum; i.e. the rate of change of flux $\frac{d \varphi_{a}}{d t}$ is zero. So, the induced voltage is zero, so the D-axis can be located on the oscillogram. When the armature m.m.f. wave is along the quadrature axis, the armature flux linkage of field winding is zero,i.e. $\frac{d \varphi_{a}}{d t}$ is maximum. Thus the Qaxis can also be located on the oscillogram.


Slip Test Connection Diagram

If oscillogram is not available, then an ammeter and a voltmeter is used as shown in the above figure. The prime mover speed is adjusted until the ammeter and voltmeter pointers swing slowly between maximum and minimum values. The maximum and minimum readings of ammeter and voltmeter are recorded.

Since applied voltage is constant, the air gap flux should be constant. When crest of the m.m.f. wave is coincident with the D-axis, the air gar reluctance is minimum, so the magnetizing current required for the establishment of constant flux is minimum. So, armature reading is minimum, so corresponding drop in the armature circuit is minimum. The armature terminal voltage is applied voltage minus armature voltage drop. So, armature supply voltage is maximum.
$\therefore X_{d}=\frac{\text { Maximum armature terminal voltage }}{\text { minimum armature current }}$

By a similar thought process;

$$
X_{q}=\frac{\text { Minimum armature terminal voltage }}{\text { maximum armature current }}
$$

The swing of the ammeter pointer is wide, but the swing of the voltmeter pointer is small, because the impedance voltage drop in the leads and connecting wire is small. Since low voltage is applied, the values measured are unsaturated values.

While performing the experiment the slip should be small; otherwise

- Current in the damper windings will introduce large error in the measurement.
- The pointers swing at high speed, making reading noting difficult.

The applied voltage should be small, because:

- it is difficult to maintain a small slip, so that reluctance torque developed by the machine is small.
- The swing in meter reading due to impedance drops can be measured.

For larger applied voltage the reluctance torque developed due to saliency, tries to bring the rotor into synchronisim with the rotating magnetic field and it is difficult to note the small change in meter readings.

However, the inertia of the moving systems of the meters also introduces error in the reading.
So, the advantages of oscillographic method are:

- Elimination of inertia effect of voltmeter and ammeter
- The possibility of large slip speed, which allows higher terminal voltage to be applied.

In practice, there may be error in reading the oscillgram and voltmeter ammeter readings are not very reliable due to effect of inertia of the moving parts. In view of this short comings the slip test is done only to determine $X_{q} / X_{d}$ ratio.
$\frac{X_{q}}{X_{d}}=\frac{\text { Minimum armature terminal voltage }}{\text { maximum armature current }} \times \frac{\text { minimum armature current }}{\text { Maximum armature terminal voltage }}$
$X_{d}$ is determined from O.C test and S.C. test, $X_{q}$ can be determined as follows:

$$
X_{q}=\frac{X_{q}}{X_{d}} \times X_{d} \text { from O.C.and S.C.test }
$$

## $X_{d}$ from O.C and S.C. Test

O.C test is performed at synchronous speed and the O.C.C is the plot between the open circuit phase voltage and the field current.
S.C test is preferably performed at synchronous speed, but can also be performed at a speed slightly different from the synchronous speed. S.C.C is the plot between the short circuit current and the field current.


The above figures represent salient pole synchronous generator approximate phasor diagram, where armature resistance ra has been neglected.

For any power factor $\cos \theta$

$$
I_{d}=I_{a} \sin (\delta+\theta) ; \quad I_{q}=I_{a} \cos (\delta+\theta) ; \quad E_{f}=V_{t} \cos \delta+I_{d} X_{d} ; V_{t} \sin \delta=I_{q} X_{q}
$$

Under short circuit condition p.f. is almost equal to zero as $r_{a} \ll X_{d}, X_{q}$ and thus it can be assumed that $r_{a} \approx 0$

Thus:
$V_{t} \sin \delta=I_{q} X_{q}=0$, but $X_{q}$ cannot be zero, so $I_{q}=0 \quad \therefore I_{d}=I_{s c}$ and $E_{f}=I_{d} X_{d}$

$$
\therefore \quad X_{d}=\frac{\text { Open circuit voltage for a field current }}{\text { short circuit currrent for the same field current }}
$$

Type equation here.

## Salient Pole Synchronous Motor Phasor Diagram:

The motor voltage equations can be obtained from the generator voltage equation by replacing la by la. Therefore the voltage equation for a salient pole synchronous motor is,

$$
\overline{V_{t}}=\overline{E_{f}}+\overline{I_{a}} r_{a}+j \bar{I}_{d} X_{d}+j \bar{I}_{q} X_{q}
$$


-

Salient pole Synchronous Motor Phasor Diagram

In order to compute $\mathbf{E}_{\mathrm{f}}$, 'ab' is drawn perpendicular to current la. So, 'ab' can be considered as an reactance $\operatorname{drop}=\mathbf{I}_{\mathrm{a}} \mathbf{X}$. From the above figure

$$
\begin{gathered}
\angle a b c=90^{\circ}-(\theta-\delta) \text { and } \angle b a c=(\theta-\delta) \\
\therefore a c=a b \cos (\theta-\delta)=X I_{a} \cos (\theta-\delta)=X I_{q} \\
\therefore X I_{q}=X_{q} I_{q} \text { and } \therefore X=X_{q}
\end{gathered}
$$

So, angular position of $\mathbf{E}_{\mathrm{f}}$ is along $o b=\widetilde{E_{f}^{\prime}}=\overline{V_{t}}-\bar{I}_{a} r_{a}-j \bar{I}_{a} X_{q}$. So the angle $\theta-\delta$ is known and $I_{d}$ and $I_{q}$ are calculated and $\overline{V_{t}}=\overline{E_{f}}+j \bar{I}_{d}\left(X_{d}-X_{q}\right)+j \bar{I}_{a} X_{q}+\bar{I}_{a} r_{a}$

Note that the term $\bar{I}_{d}\left(X_{d}-X_{q}\right)$ appear due to saliency and reduces to zero for cylindrical rotor synchronous machines (as $X_{d}=X_{q}$ ). In salient pole machine $\mathbf{X}_{\mathbf{d}}$ is approximately $60 \%$ larger than $\mathbf{X}_{\mathbf{q}}$. However in cylindrical rotor synchronous machine $\mathbf{X}_{\mathbf{d}}$ and $\mathbf{X}_{\mathbf{q}}$ may differ slightly due to the effect of field winding slots in $q$-axis.

## Problem 1:

A salient pole synchronous generator has the following per unit parameters: $\mathbf{X}_{\mathbf{d}}=1.2 ; \mathbf{X}_{\mathbf{q}}=0.8 ; \mathbf{r}_{\mathrm{a}}=0.025$
Compute the excitation voltage on a per unit basis, when the generator is delivering rated kVA at rated voltage and at a power factor (a) o. 8 lagging and (b) 0.8 leading

## Solution:

(a) $\quad \bar{V}_{t}=1.00+j 0.00$ and $\bar{I}_{a}=1.00 \angle-36.9^{0}=0.80-j 0.60$

$$
\begin{gathered}
j I_{a} X_{q}=j(0.80-j 0.60) 0.80=0.48+\mathrm{j} 0.64 \text { and } \bar{I}_{a} r_{a}=0.020-j 0.015 \\
\widetilde{E_{f}^{\prime}}=\bar{V}_{t}+\bar{I}_{a} r_{a}+j \bar{I}_{a} X_{q}=1.00+j 0.00+0.48+\mathrm{j} 0.64+0.020-j 0.015 \\
=1.50+j 0.625=1.625 \angle 22.6^{\circ}
\end{gathered}
$$

$$
\therefore \delta=22.6^{\circ} \text { and } \angle \delta+\theta=22.6^{\circ}+36.9^{0}=59.5^{\circ}
$$

$$
\therefore I_{d}=1.00 \sin 59.5^{\circ}=0.861 \quad \text { and } \quad I_{q}=1.00 \cos 59.5^{0}=0.507
$$

$$
E_{f}=E_{f}^{\prime}+I_{d}\left(X_{d}-X_{q}\right)=1.625+0.861 \times 0.4=1.9694
$$

$$
\overline{E_{f}}=1.9694 \angle 22.6^{\circ}
$$

(b) $\quad \bar{V}_{t}=1.00+j 0.00$ and $\bar{I}_{a}=1.00 \angle 36.9^{0}=0.80+j 0.60$

$$
\begin{gathered}
j I_{a} X_{q}=j(0.80+j 0.60) 0.80=-0.48+\mathrm{j} 0.64 \text { and } \bar{I}_{a} r_{a}=0.020+j 0.015 \\
\widetilde{E_{f}^{\prime}}=\bar{V}_{t}+\bar{I}_{a} r_{a}+j \bar{I}_{a} X_{q}=1.00+j 0.00-0.48+\mathrm{j} 0.64+0.020+j 0.015 \\
=0.54+j 0.655=0.849 \angle 50.50^{\circ} \\
\therefore \delta=50.50^{\circ} \text { and } \angle \delta-\theta=50.50^{\circ}-36.9^{0}=13.60^{0} \\
\therefore I_{d}=1.00 \sin 13.60^{0^{0}}=0.235 \quad \text { and } \quad I_{q}=1.00 \cos 13.60^{00}=0.9719 \\
E_{f}=E_{f}^{\prime}+I_{d}\left(X_{d}-X_{q}\right)=0.849+0.235 \times 0.4=0.943 \\
\overline{E_{f}}=0.943 \angle 50.50^{\circ}
\end{gathered}
$$

## Problem 2:

For an over excited Synchronous motor, prove that:

$$
\begin{aligned}
\tan \delta & =\frac{I_{a}\left(X_{q} \cos \theta+r_{a} \sin \theta\right)}{V_{t}+I_{a}\left(X_{q} \sin \theta-r_{a} \cos \theta\right)} \\
X_{q} & =\frac{V_{t} \sin \delta-I_{a} r_{a} \sin (\delta+\theta)}{I_{a} \cos (\delta+\theta)}
\end{aligned}
$$

## Solution:

$$
\begin{gathered}
\mathrm{ab}=\mathrm{ac}+\mathrm{cd}=I_{d} r_{a}+I_{q} X_{q} \\
\sin \delta=\frac{a b}{o a}=\frac{a c+a d}{o a}=\frac{I_{d} r_{a}+I_{q} X_{q}}{V_{t}} \\
\text { or } V_{t} \sin \delta=I_{d} r_{a}+I_{q} X_{q}
\end{gathered}
$$

$$
\text { Again, } I_{d}=I_{a} \sin (\delta+\theta) \quad \text { and } I_{q}=I_{a} \cos (\delta+\theta)
$$

$$
\therefore \quad V_{t} \sin \delta=I_{d} r_{a}+I_{q} X_{q}=I_{a} r_{a} \sin (\delta+\theta)+I_{a} \mathrm{X}_{\mathrm{q}} \cos (\delta+\theta)
$$

$$
=I_{a} r_{a}(\sin \delta \cos \theta+\cos \delta \sin \theta)
$$

$$
+I_{a} \mathrm{X}_{\mathrm{q}}(\cos \delta \cos \theta-\sin \delta \sin \theta)
$$

$$
\sin \delta\left(V_{t}-I_{a} r_{a} \cos \theta+I_{a} X_{q} \sin \theta\right)=\cos \delta\left(I_{a} r_{a} \sin \theta+I_{a} X_{q} \cos \theta\right)
$$



$$
\begin{gathered}
\therefore \tan \delta=\frac{\sin \delta}{\cos \delta} \\
=\frac{\left(I_{a} r_{a} \sin \theta+I_{a} X_{q} \cos \theta\right)}{\left(V_{t}-I_{a} r_{a} \cos \theta+I_{a} X_{q} \sin \theta\right)} \\
=\frac{I_{a}\left(r_{a} \sin \theta+X_{q} \cos \theta\right)}{\left(V_{t}-I_{a}\left(r_{a} \cos \theta+X_{q} \sin \theta\right)\right)}
\end{gathered}
$$

$$
V_{t} \sin \delta=I_{d} r_{a}+I_{q} X_{q}
$$

$$
\therefore X_{q}=\frac{V_{t} \sin \delta-I_{d} r_{a}}{I_{q}}
$$

$$
X_{q}=\frac{V_{t} \sin \delta-I_{a} \sin (\delta+\theta) r_{a}}{I_{a} \cos (\delta+\theta)}
$$

Power Angle Characteristics of a Salient Pole Synchronous Machine:


From the diagram per phase power developed is:

$$
\begin{gathered}
P=I_{d} V_{d}+I_{q} V_{q}=I_{d} V_{t} \sin \delta+I_{q} V_{t} \cos \delta \\
V_{t} \sin \delta=a b=d c=I_{q} X_{q} \therefore I_{q}=\frac{V_{t} \sin \delta}{X_{q}} \\
V_{t} \cos \delta=o a=o d-a d=o d-b c \\
=E_{f}-I_{d} X_{d} \\
\therefore I_{d}=\frac{E_{f}-V_{t} \cos \delta}{X_{d}}
\end{gathered}
$$

$$
\begin{aligned}
& \quad \therefore P=I_{d} V_{t} \sin \delta+I_{q} V_{t} \cos \delta \\
& =\frac{E_{f}-V_{t} \cos \delta}{X_{d}} V_{t} \sin \delta+\frac{V_{t} \sin \delta}{X_{q}} V_{t} \cos \delta \\
& =\frac{E_{f} V_{t} \sin \delta}{X_{d}}+V_{t}^{2}\left(\frac{1}{X_{q}}-\frac{1}{X_{d}}\right) \sin \delta \cos \delta \\
& \therefore P=\frac{E_{f} V_{t} \sin \delta}{X_{d}}+\frac{V_{t}^{2}}{2}\left(\frac{1}{X_{q}}-\frac{1}{X_{d}}\right) \sin 2 \delta
\end{aligned}
$$

So, power expression has two components. The first term is called electromagnetic power, because its existence depends on the presence of field excitation. This is similar to the power expression of a cylindrical rotor machine.
The second term is called reluctance power. Reluctance power exists even when the field excitation is zero. In a salient pole machine the reluctance along the direct axis and the quadrature axis are different and the armature reaction has a tendency to get oriented along the low reluctance path i.e. along the direct axis and the reluctance power is developed. Since $\frac{V_{t}^{2}}{2}\left(\frac{1}{X_{q}}-\frac{1}{X_{d}}\right) \sin 2 \delta$ because of the difference in reluctance, it is called reluctance power and $\frac{1}{\omega_{s}} \frac{V_{t}^{2}}{2}\left(\frac{1}{X_{q}}-\frac{1}{X_{d}}\right) \sin 2 \delta$ is called reluctance torque. So, a salient pole motor connected to the infinite bus will run as a reluctance motor when its field current is reduced to zero.
For maximum power $\frac{d P}{d \delta}=0$

$$
\begin{aligned}
& \therefore \frac{d P}{d \delta}=\frac{E_{f} V_{t} \cos \delta}{X_{d}}+V_{t}^{2}\left(\frac{1}{X_{q}}-\frac{1}{X_{d}}\right) \cos 2 \delta=0 \\
& \therefore \cos \delta=-\frac{E_{f} X_{q}}{4 V_{t}\left(X_{d}-X_{q}\right)} \pm \sqrt{\frac{1}{2}+\left[\frac{E_{f} X_{q}}{4 V_{t}\left(X_{d}-X_{q}\right)}\right]^{2}}
\end{aligned}
$$

## Synchronizing Power and Synchronizing Torque:

The variation of synchronous power associated with small change in load angle $\delta$, is called synchronizing power. Synchronizing power coefficient $\mathrm{P}_{\text {sy }}$, given by:

$$
P_{s y}=\frac{d P}{d \delta}=\frac{E_{f} V_{t}}{X_{s}} \cos \delta+V_{t}^{2}\left(\frac{1}{X_{q}}-\frac{1}{X_{d}}\right) \cos 2 \delta
$$



Power Angle Characteristics of Salient Pole Machine

So for small change in load angle synchronizing power is given by :

$$
P_{s}=\frac{d P}{d \delta} \Delta \delta=\frac{E_{f} V_{t}}{X_{s}} \cos \delta \Delta \delta+V_{t}^{2}\left(\frac{1}{X_{q}}-\frac{1}{X_{d}}\right) \cos 2 \delta \Delta \delta
$$

Synchronizing torque is given by $T_{s}=\frac{m P_{s}}{\omega_{s}} ; \quad$ or, $T_{s}=\frac{1}{\omega_{s}} m \frac{d P}{d \delta} \Delta \delta$;

$$
\text { or } T_{s}=m \frac{1}{\omega_{s}}\left(\frac{E_{f} V_{t}}{X_{s}} \cos \delta \Delta \delta+V_{t}^{2}\left(\frac{1}{X_{q}}-\frac{1}{X_{d}}\right) \cos 2 \delta \Delta \delta\right)
$$

