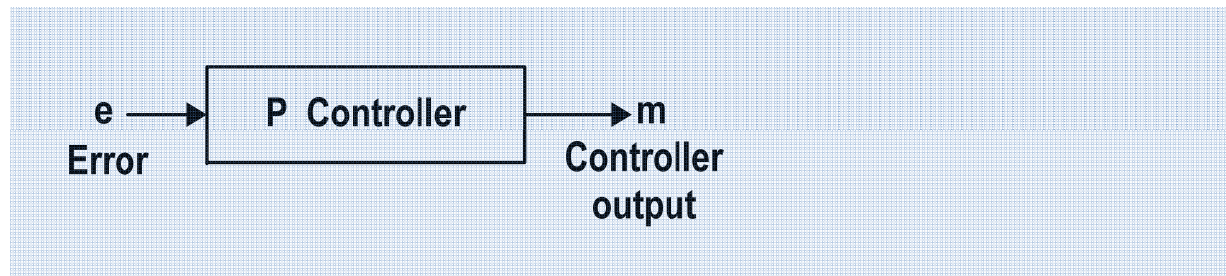


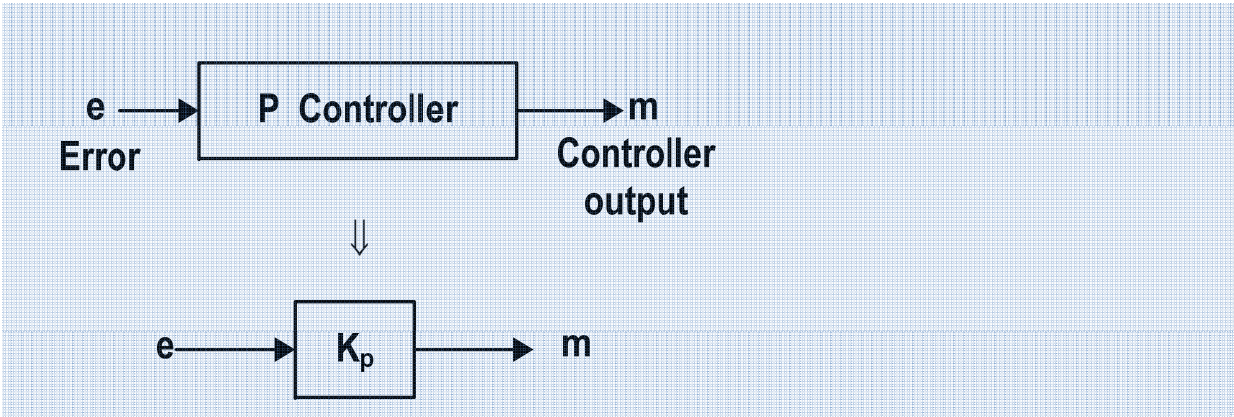
Process Controllers

Prof. Anjan Rakshit and Dr. Amitava Chatterjee
Electrical Measurements and Instrumentation
Laboratory,
Electrical Engineering Department,
Jadavpur University, Kolkata, India.

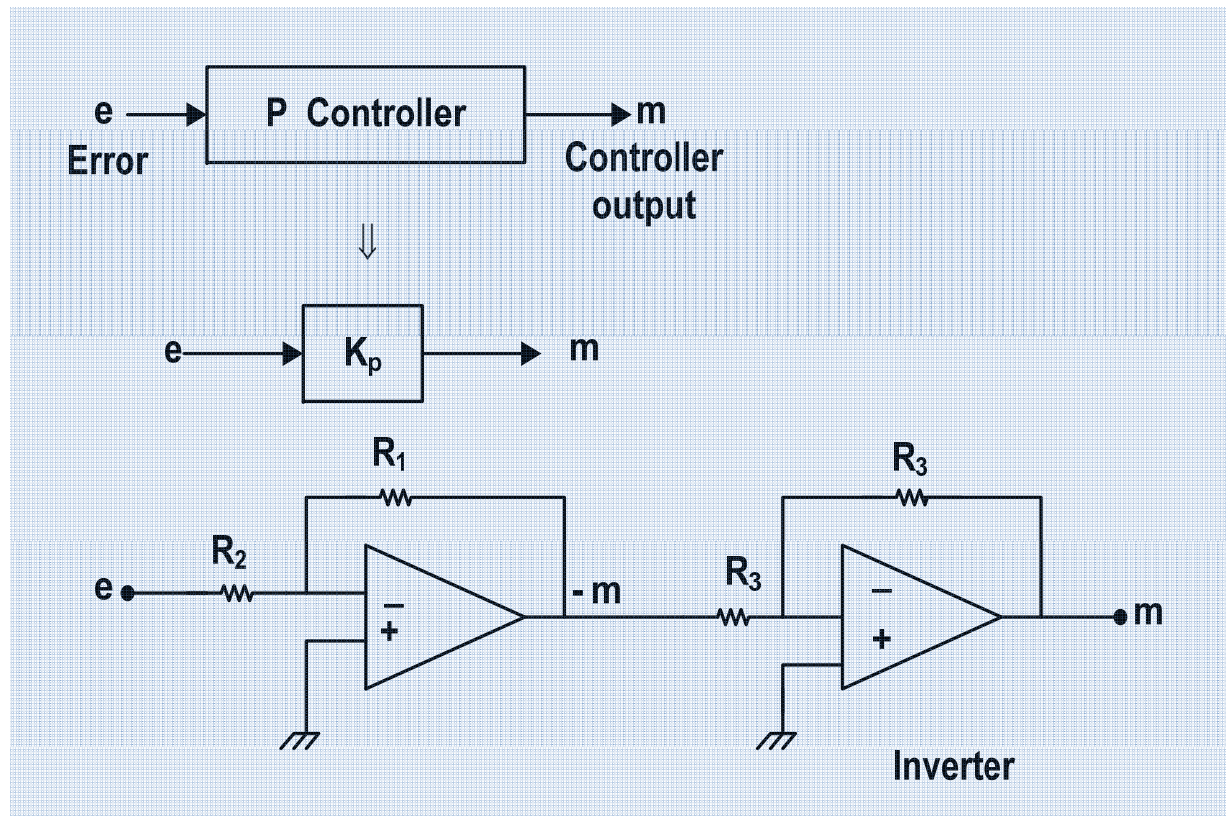
Realization of Electronic Proportional (P) controller



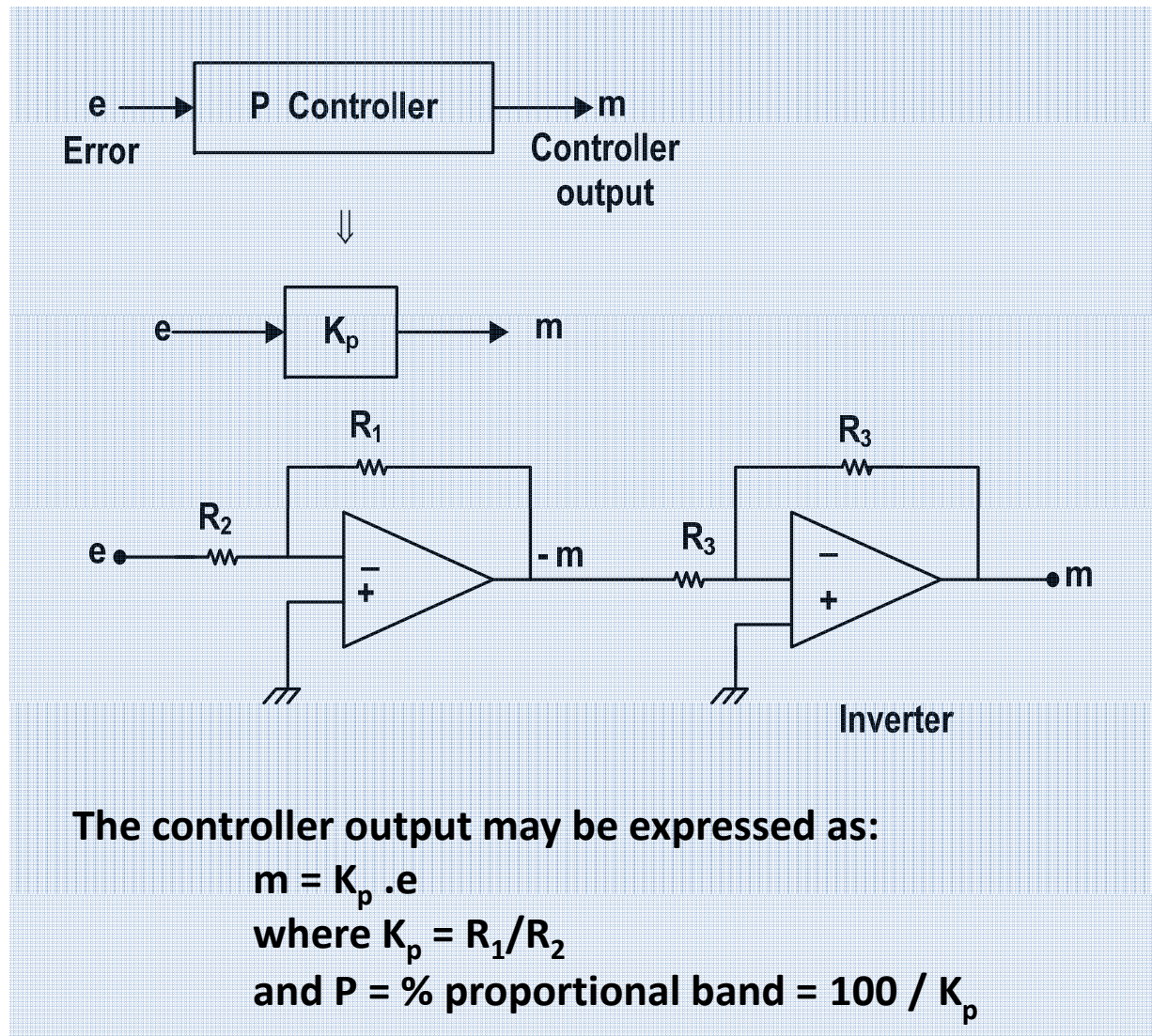
Realization of Electronic Proportional (P) controller



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Realization of Electronic PID controller (P, PI, & PD are special cases of PID)

The controller output may be expressed as:

$$m = K_p \left(e + \frac{1}{T_i} \int_0^t e dt + T_D \frac{de}{dt} \right)$$

where T_i = integral time constant
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Or,

$$m = K_p e + \frac{K_p}{T_i} \int_0^t e dt + K_p T_D \frac{de}{dt}$$

Taking Laplace transform with zero initial conditions,

$$M(s) = K_p E(s) + \frac{K_p E(s)}{sT_i} + sK_p T_D E(s)$$

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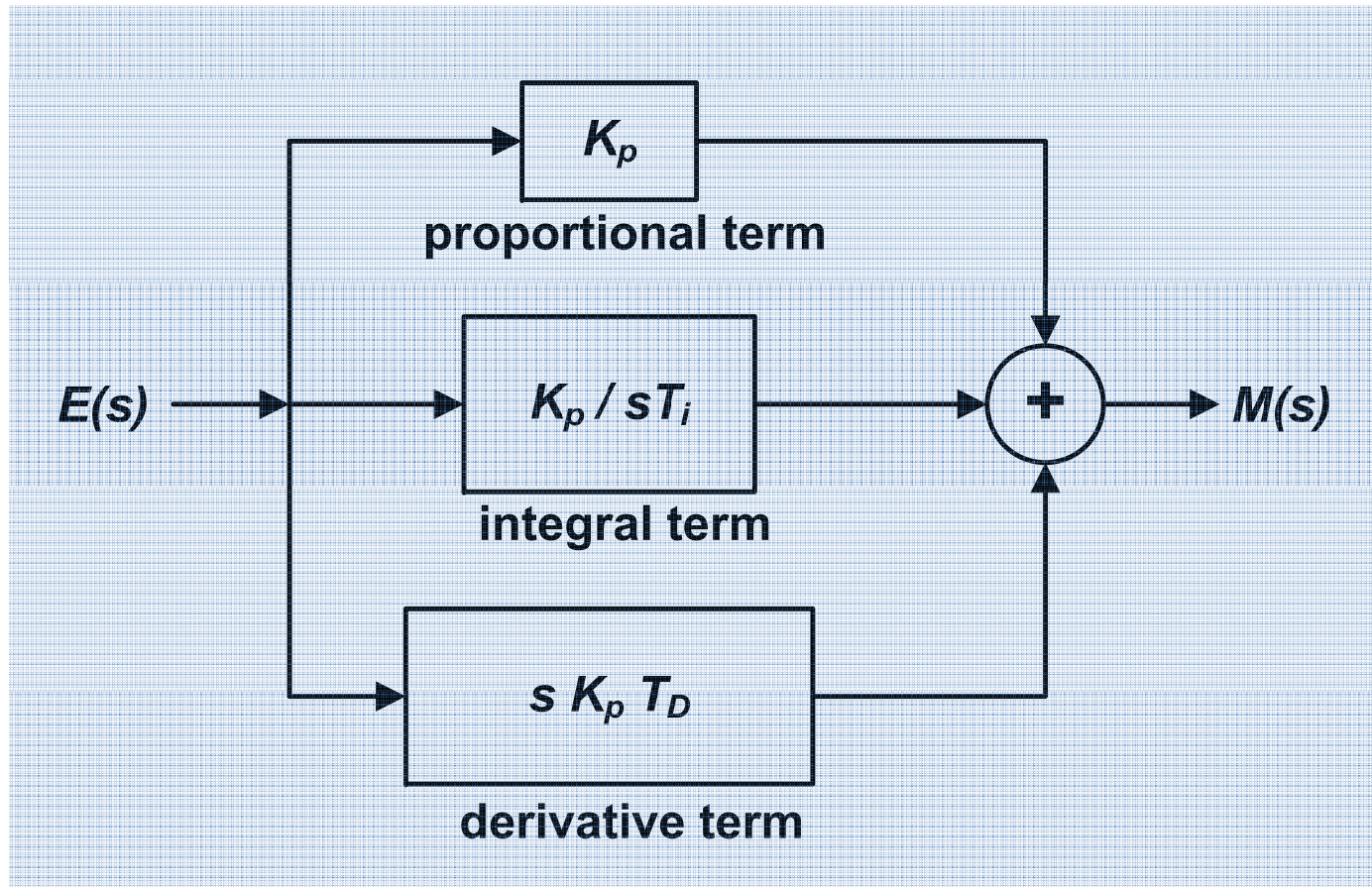
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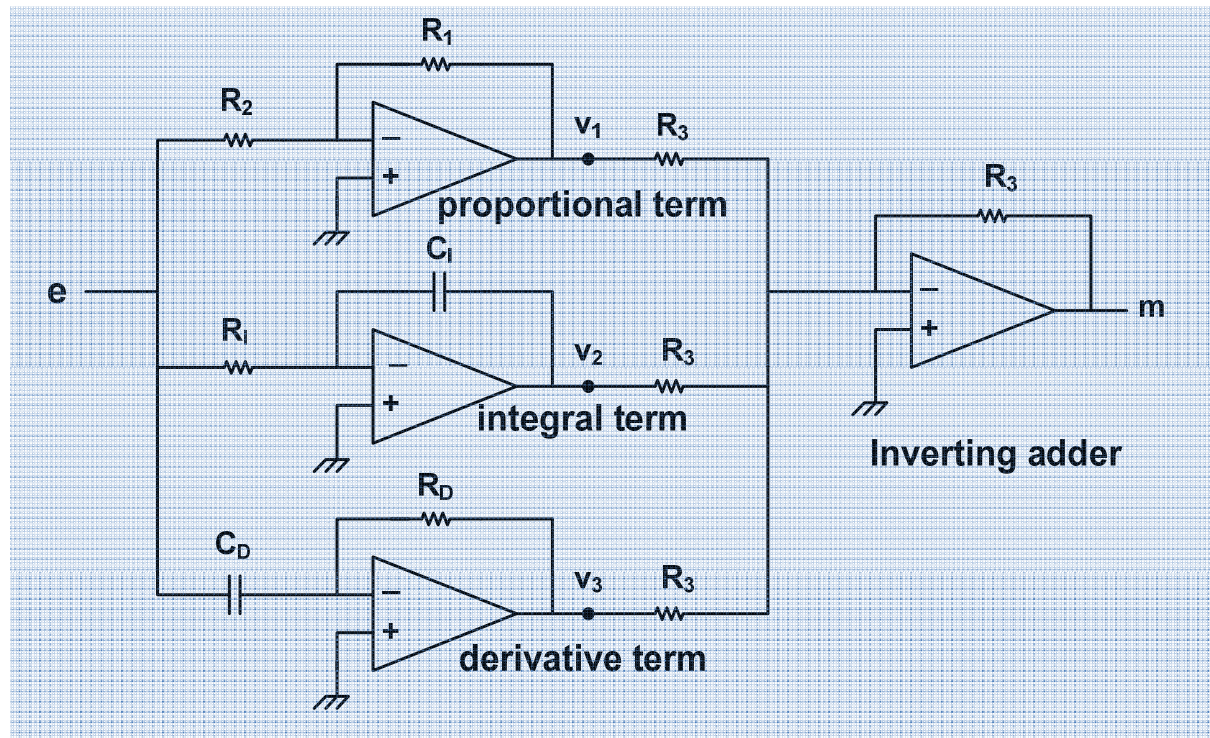
$$\frac{M(s)}{E(s)} = K_p \left(1 + \frac{1}{sT_i} + sT_D \right)$$

the transfer function of PID controller

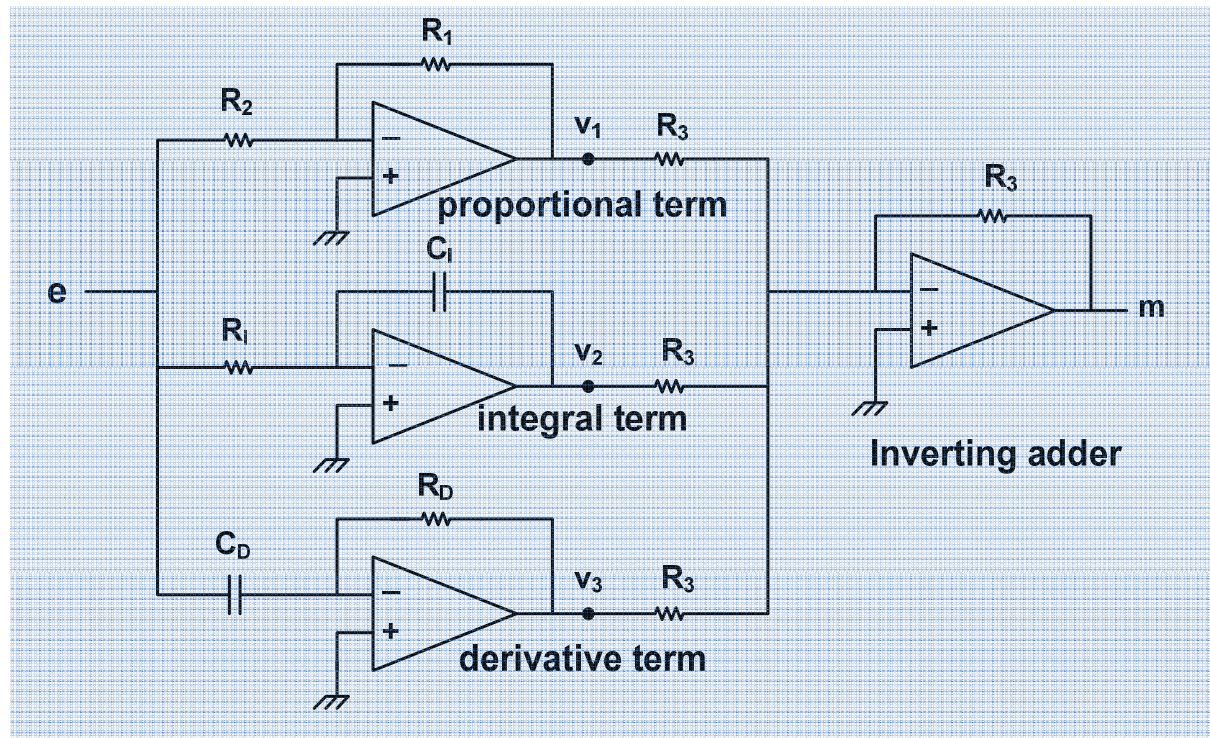
Parallel Realization of PID Controller



Circuit realization

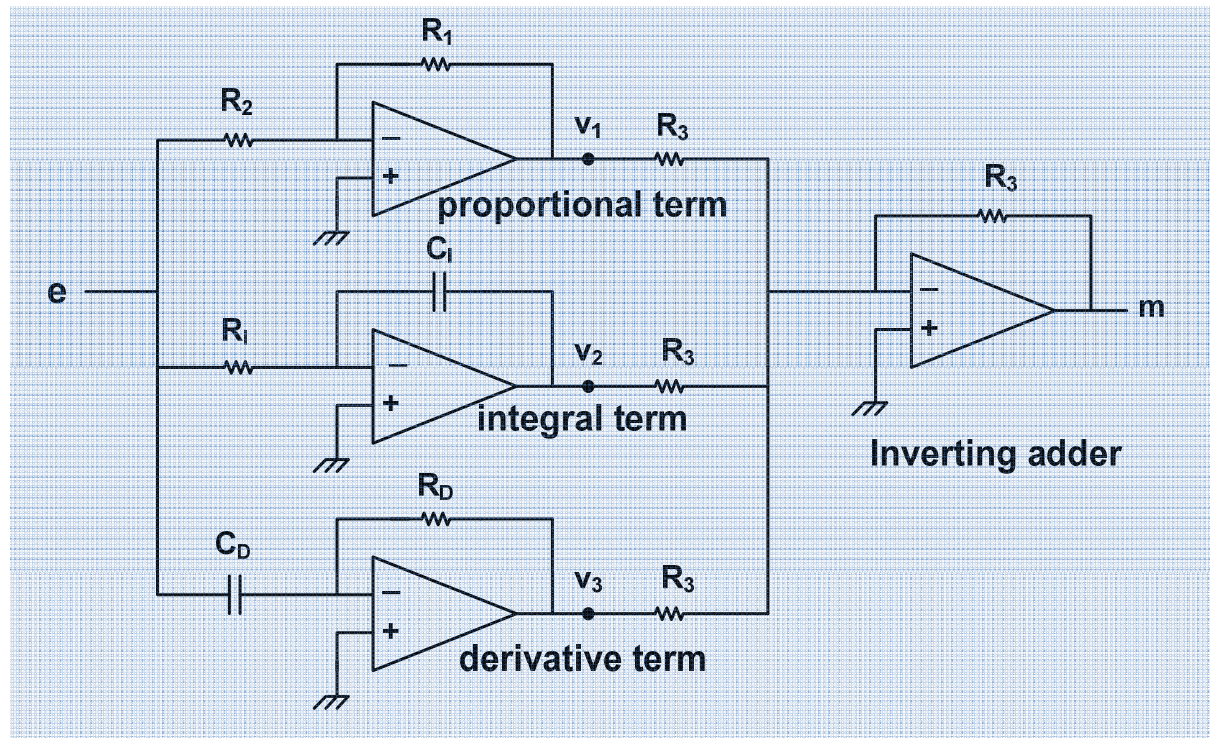


Circuit realization



$$M(s) = -(V_1(s) + V_2(s) + V_3(s))$$

Circuit realization



$$M(s) = -(V_1(s) + V_2(s) + V_3(s))$$

Or,

$$M(s) = E(s) \left[\frac{R_1}{R_2} + \frac{1}{sR_I C_I} + sR_D C_D \right]$$

Circuit realization

Therefore, transfer function of PID controller becomes

$$\frac{M(s)}{E(s)} = \left[\frac{R_1}{R_2} + \frac{1}{sR_I C_I} + sR_D C_D \right]$$

Here,

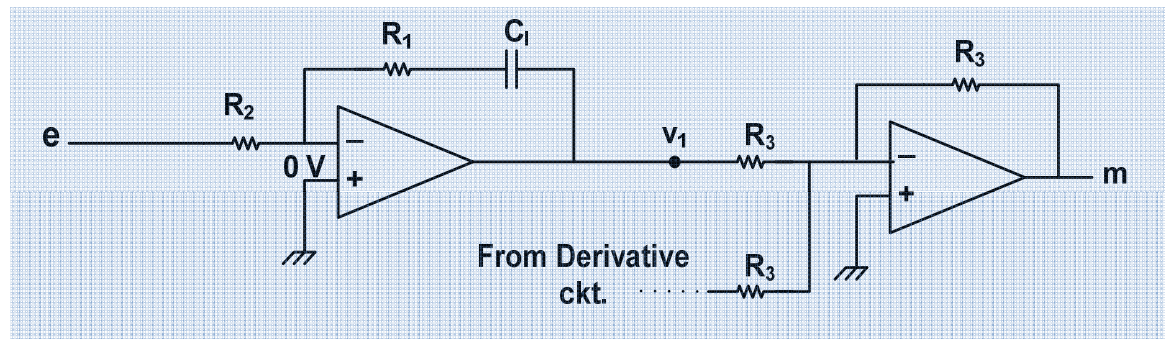
$$K_p = \frac{R_1}{R_2}$$

$$\frac{K_p}{T_i} = \frac{1}{R_I C_I}$$

$$K_p T_D = R_D C_D$$

Circuit realization

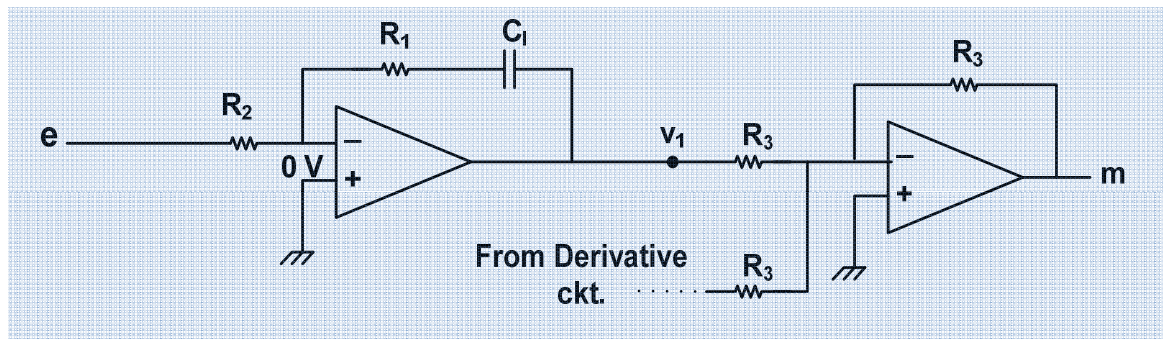
Proportional and integral terms may be combined as:



$$\frac{E(s)}{R_2} + \frac{V_1(s)}{R_1 + \frac{1}{sC_1}} = 0$$

Circuit realization

Proportional and integral terms may be combined as:



$$\frac{E(s)}{R_2} + \frac{V_1(s)}{R_1 + \frac{1}{sC_I}} = 0$$

$$\text{Or, } \frac{E(s)}{R_2} = -\frac{V_1(s)}{R_1 + \frac{1}{sC_I}} = -\frac{V_1(s)sC_I}{1 + sR_1C_I}$$

Circuit realization

$$\text{Or, } \frac{V_1(s)}{E(s)} = - \frac{1 + sR_1C_I}{sR_2C_I}$$

Circuit realization

$$\begin{aligned}\text{Or, } \frac{V_1(s)}{E(s)} &= - \frac{1 + sR_1C_I}{sR_2C_I} \\ &= - \left(\frac{R_1}{R_2} + \frac{1}{sR_2C_I} \right)\end{aligned}$$

Circuit realization

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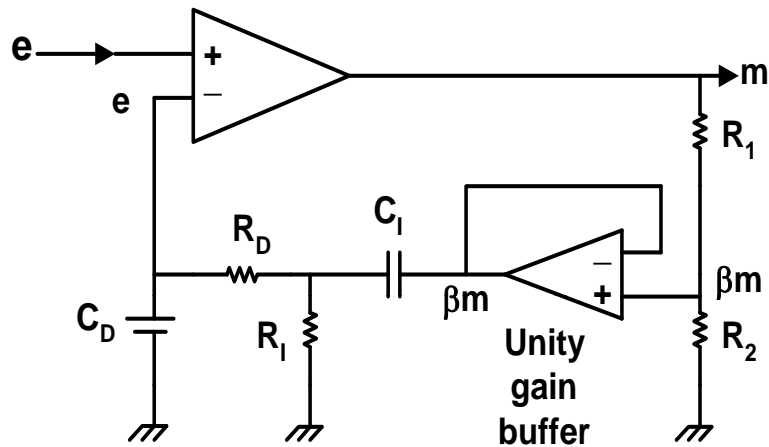
Circuit realization

$$\begin{aligned}\text{Or, } \frac{V_1(s)}{E(s)} &= - \frac{1 + sR_1C_I}{sR_2C_I} \\ &= - \left(\frac{R_1}{R_2} + \frac{1}{sR_2C_I} \right) \\ &= - \frac{R_1}{R_2} \left(1 + \frac{1}{sR_1C_I} \right) \\ &= - K_p \left(1 + \frac{1}{sT_i} \right)\end{aligned}$$

Where

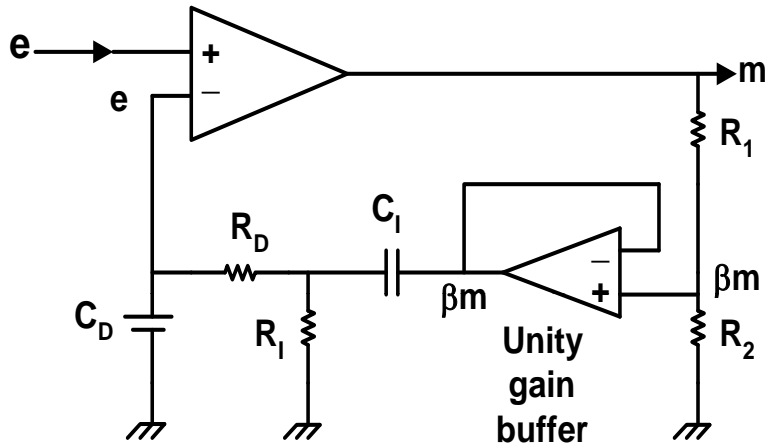
$$K_p = \frac{R_1}{R_2} \quad \text{and} \quad T_i = R_1C_I$$

A simple PID Controller with two op-amps

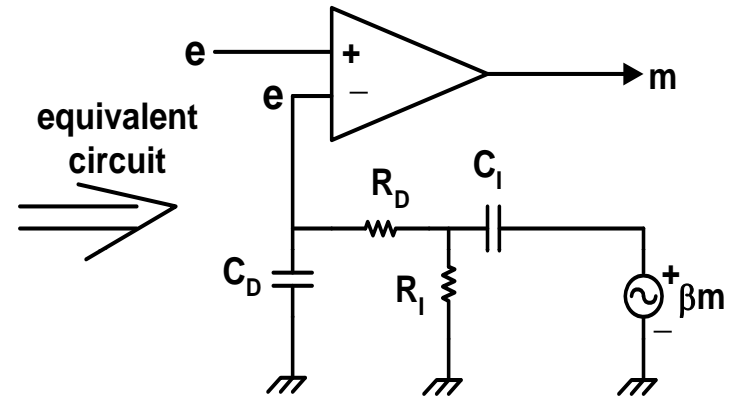


Here,
$$\beta = \frac{R_2}{R_1 + R_2}$$

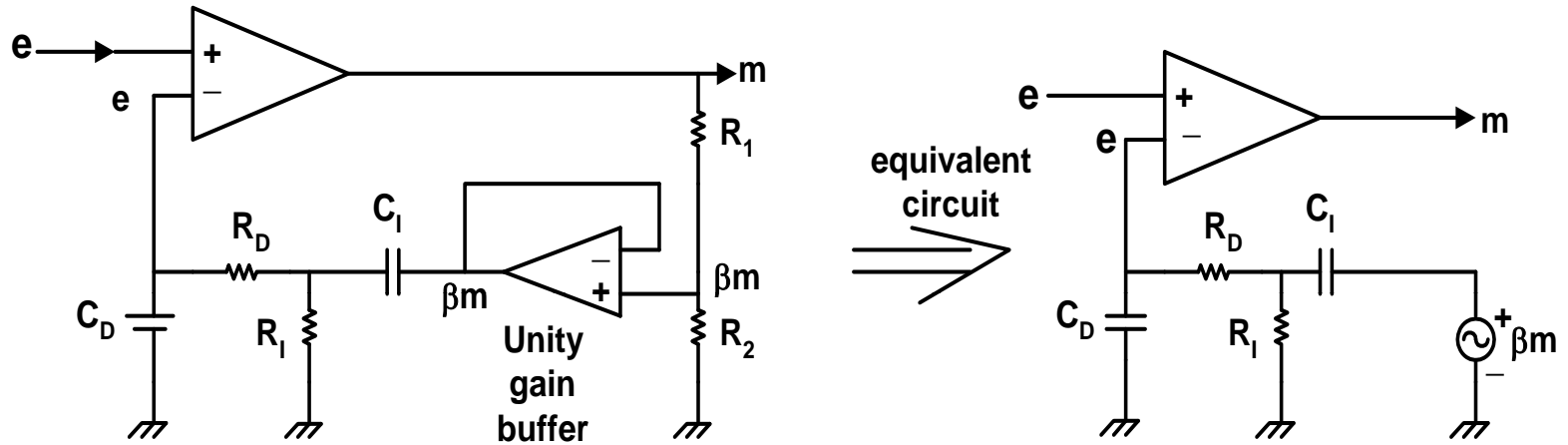
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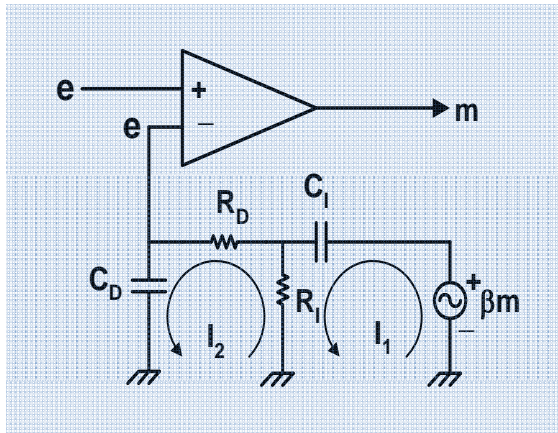
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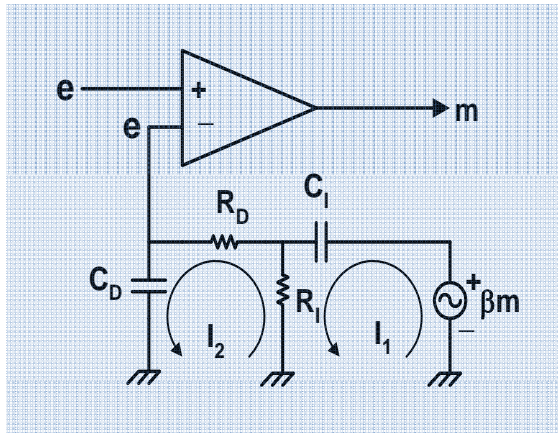
Here,
$$\beta = \frac{R_2}{R_1 + R_2}$$

The unity gain buffer amplifier is required to avoid the loading effect of the feedback network

Circuit solution



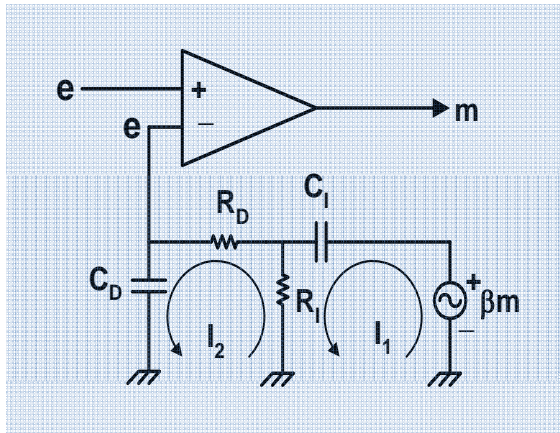
Circuit solution



Loop equations:
$$\beta M(s) = I_1 \left(R_I + \frac{1}{sC_I} \right) - I_2 R_I$$

$$0 = -I_1 R_I + I_2 \left(R_I + R_D + \frac{1}{sC_D} \right)$$

Circuit solution



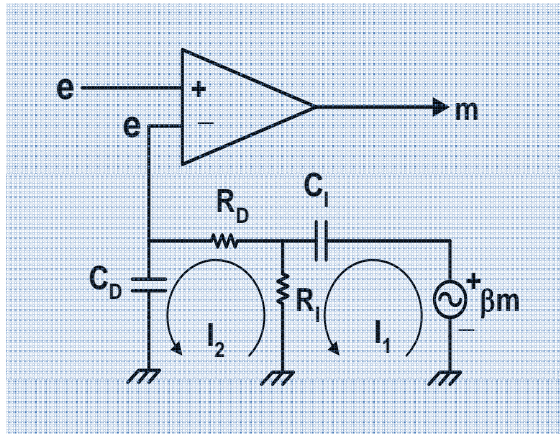
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In matrix form:

$$\begin{bmatrix} R_I + \frac{1}{sC_I} & -R_I \\ -R_I & R_I + R_D + \frac{1}{sC_D} \end{bmatrix} \begin{bmatrix} I_1 \\ I_2 \end{bmatrix} = \begin{bmatrix} \beta M(s) \\ 0 \end{bmatrix}$$

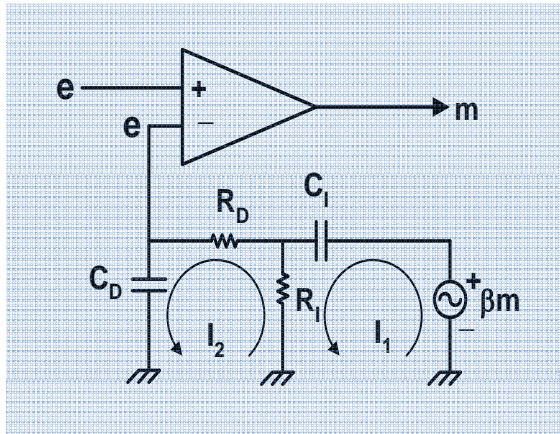
Circuit solution



By Cramer's rule:

$$I_2 = \frac{\begin{vmatrix} R_I + \frac{1}{sC_I} & \beta M(s) \\ -R_I & 0 \end{vmatrix}}{\begin{vmatrix} R_I + \frac{1}{sC_I} & -R_I \\ -R_I & R_I + R_D + \frac{1}{sC_D} \end{vmatrix}}$$

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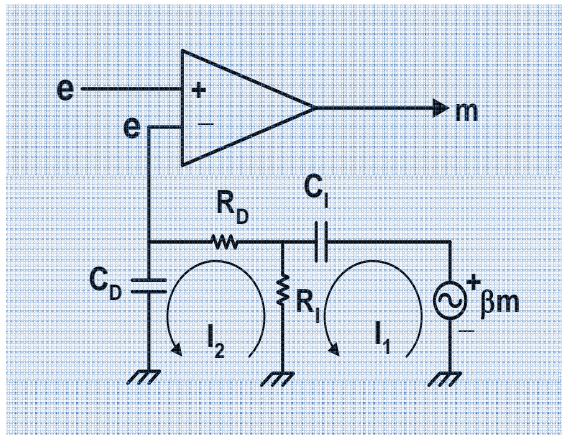


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$$\text{Or, } I_2 = \frac{\beta M(s) R_I}{\left(R_I + \frac{1}{sC_I} \right) \left(R_I + R_D + \frac{1}{sC_D} \right) - R_I^2}$$

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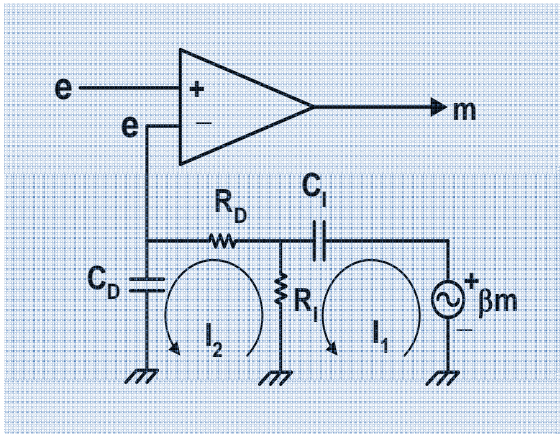
$$\text{Or, } I_2 = \frac{\beta M(s) R_I}{\left(R_I + \frac{1}{sC_I} \right) \left(R_I + R_D + \frac{1}{sC_D} \right) - R_I^2}$$

$$\therefore E(s) = \frac{I_2}{sC_D} = \frac{\beta M(s) R_I / (sC_D)}{\left(R_I + \frac{1}{sC_I} \right) \left(R_I + R_D + \frac{1}{sC_D} \right) - R_I^2}$$

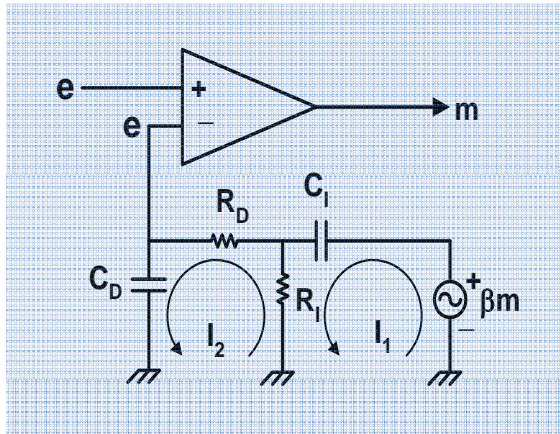
Circuit solution

Then, the T.F. becomes:

$$\frac{M(s)}{E(s)} = \frac{1}{\beta} \left[\frac{\left\{ \left(R_I + \frac{1}{sC_I} \right) \left(R_I + R_D + \frac{1}{sC_D} \right) - R_I^2 \right\} sC_D}{R_I} \right]$$



Circuit solution



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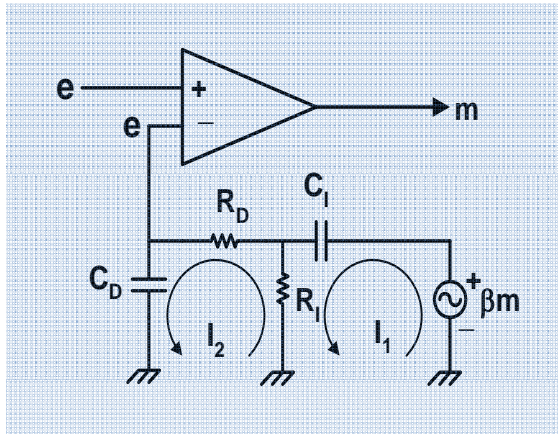
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Simplifying,

$$\frac{M(s)}{E(s)} = \frac{1}{\beta} \left[\left(\frac{T_1 + T_2 + R_I C_D}{T_1} \right) + \frac{1}{sT_1} + sT_2 \right]$$

where $T_1 = R_I C_I$ and $T_2 = R_D C_D$

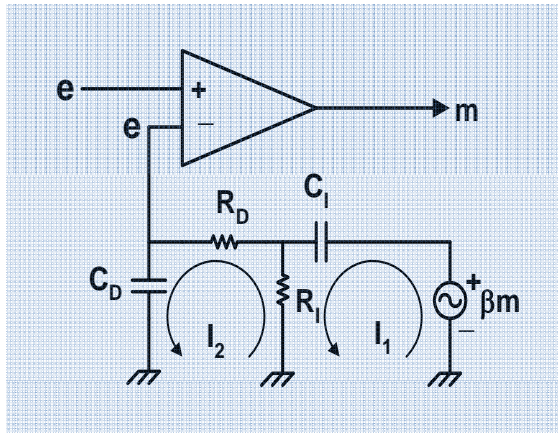
Circuit solution



By substituting
$$A = \frac{T_1 + T_2 + R_I C_D}{T_1}$$

Or,
$$A = 1 + \frac{C_D}{C_I} + \frac{T_2}{T_1}$$

Circuit solution

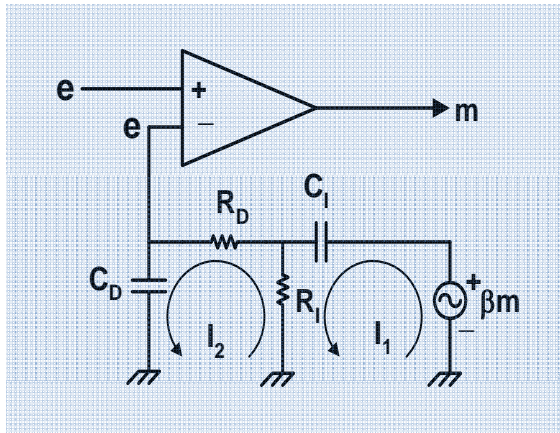


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This expression of A presents the problem of interaction and hence, the controller developed is called an **interacting controller**.

Circuit solution



By substituting $A = \frac{T_1 + T_2 + R_I C_D}{T_1}$

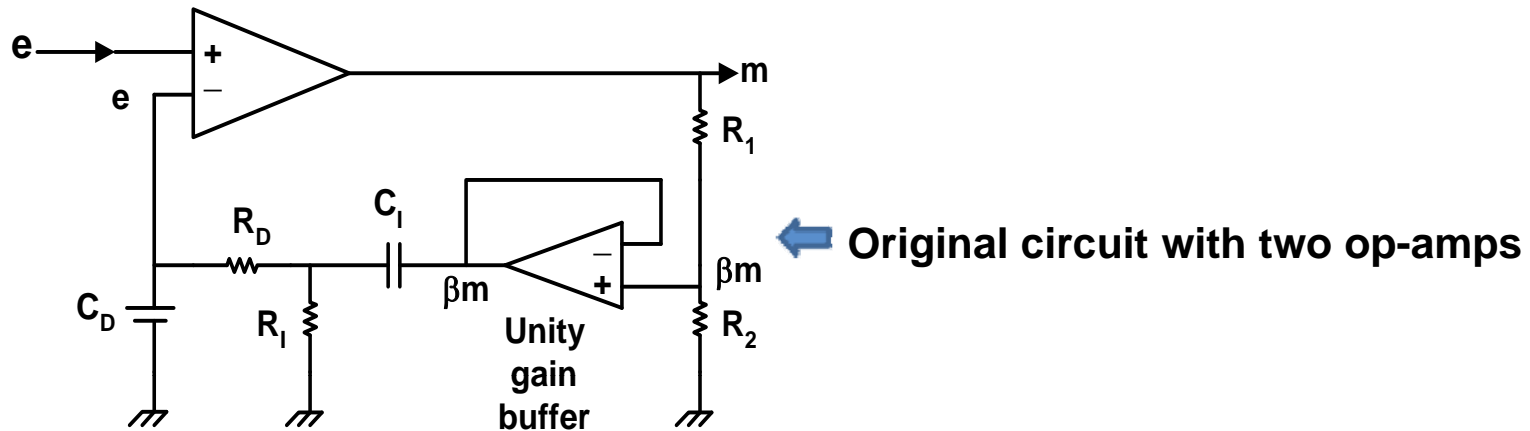
Or, $A = 1 + \frac{C_D}{C_I} + \frac{T_2}{T_1}$

The T.F. becomes:

$$\begin{aligned} \frac{M(s)}{E(s)} &= \frac{A}{\beta} \left[1 + \frac{1}{sT_1 A} + \frac{sT_2}{A} \right] \\ &= K_p \left[1 + \frac{1}{sT_I} + sT_D \right] \end{aligned}$$

Where $K_p = \frac{A}{\beta}$, $T_I = T_1 A$, and $T_D = T_2/A$

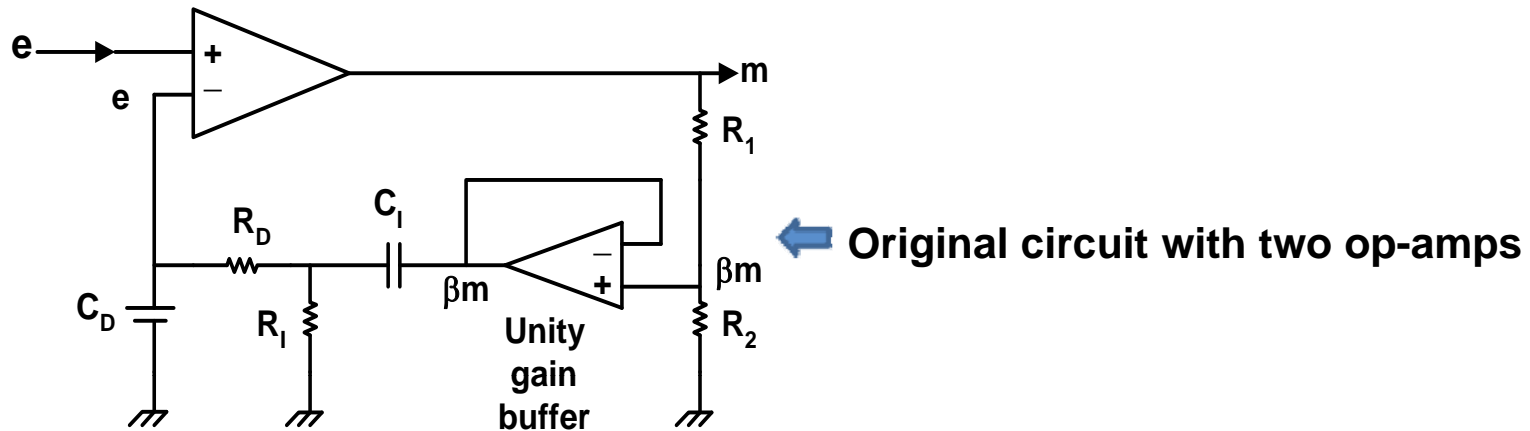
A simplified PID Controller with one op-amp



The unity gain buffer amplifier may be omitted if the resistances are chosen such that

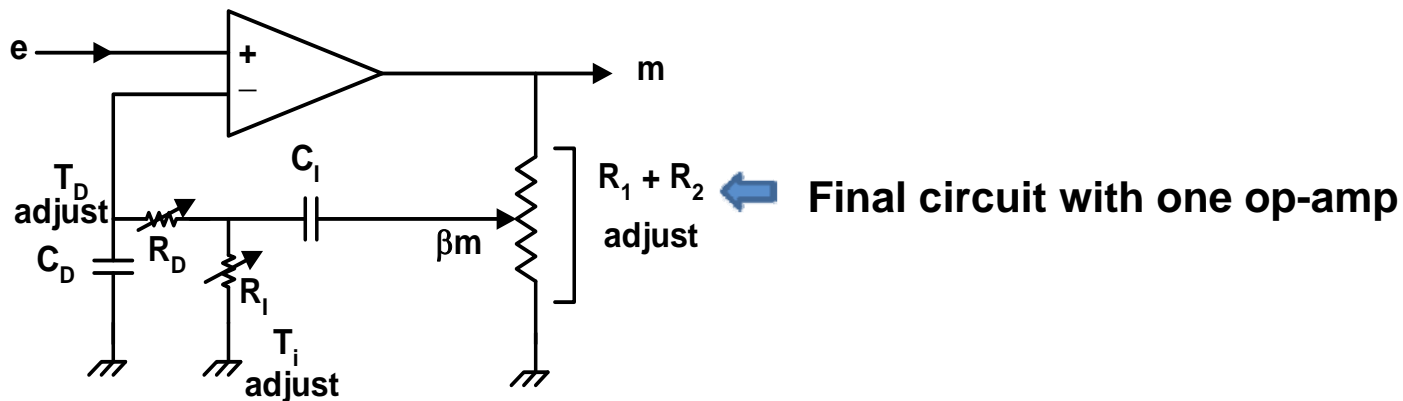
$$(R_I \parallel R_D) \gg (R_1 \parallel R_2)$$

A simplified PID Controller with one op-amp



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- Provision for providing a **bias** term

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- Provision for **Local/Remote** modes of operation

Provision for providing a bias term

The controller output is $m = K_p \left(e + \frac{1}{T_i} \int_0^t e dt + T_D \frac{de}{dt} \right) + \text{bias}$

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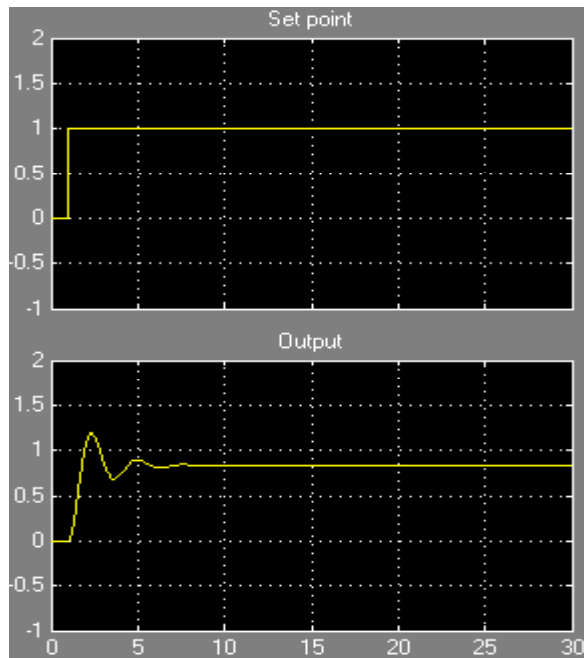
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Bias may be properly adjusted for zero steady state error with P and PD controllers, when load is constant. Proper bias can also ensure efficient operation under start-up condition.

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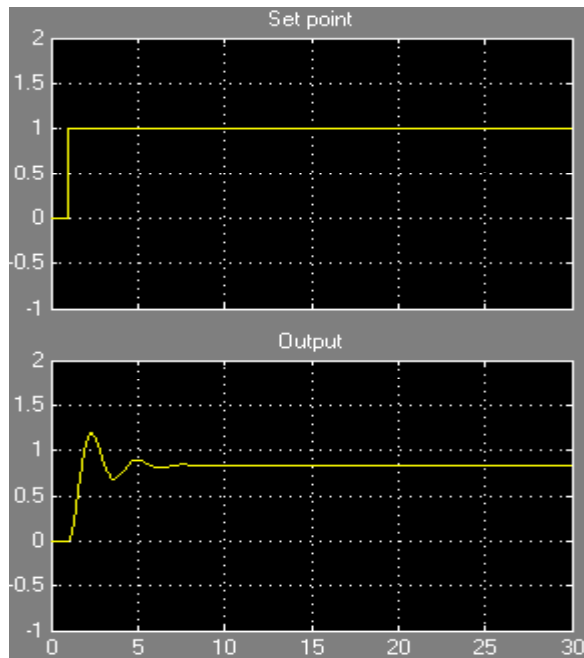
← Non-zero steady state error

P-control without bias

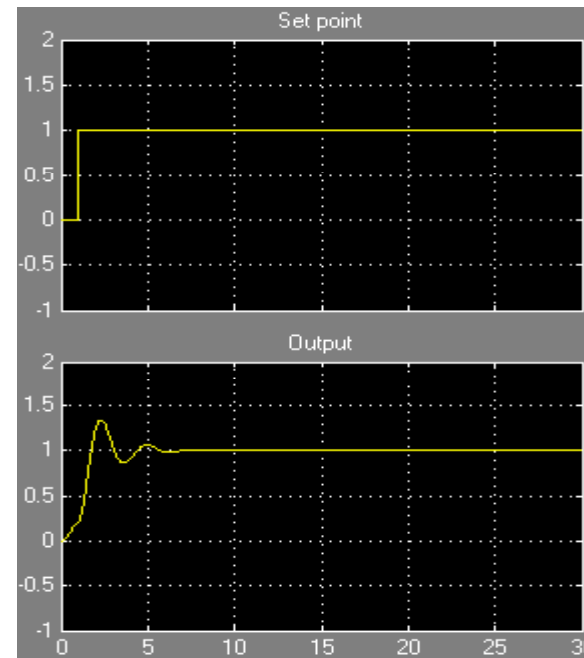
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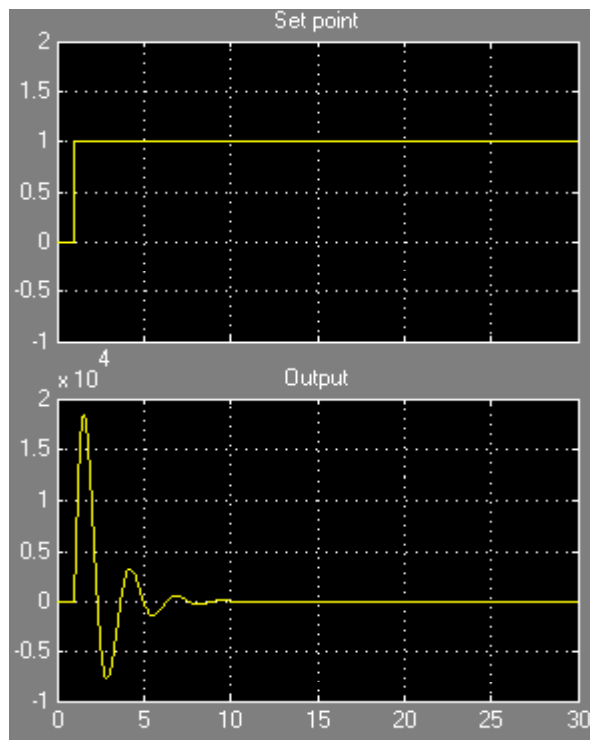
P-control with bias

Provision for anti-derivative kick

Derivative action may produce an unwanted kick at the controller output when there is a step change in the set-point. This effect may be eliminated if the derivative term is computed from the measured variable, instead of the error (anti-derivative kick feature).

Provision for anti-derivative kick

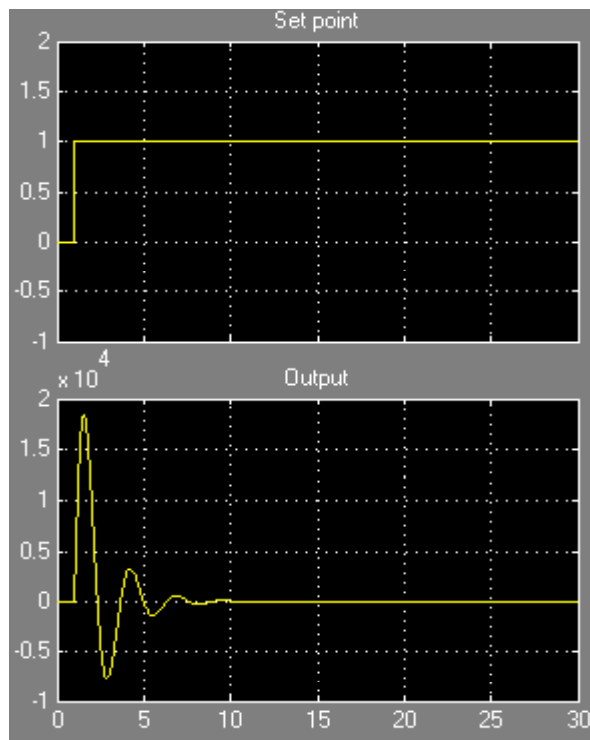
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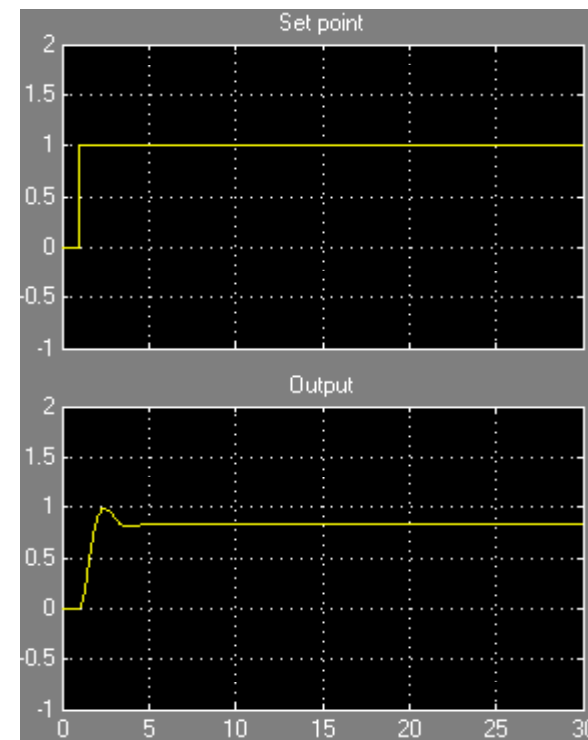
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PD-control with derivative kick



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Integral action in PI and PID controllers may produce a large integral error term when a non-zero error persists for a long time. Integration action may be switched off to combat this situation, otherwise, a long time may be required to come back to normal working range.

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In some controllers, when the integral term is present, the integral amount during T_i is summed with each pass through the calculation and becomes the controller bias. This technique is known as automatic reset and this is done to avoid long duration operation of the integral action – thus error due to drift etc. during integration may be avoided.

Output must be limited (saturation)

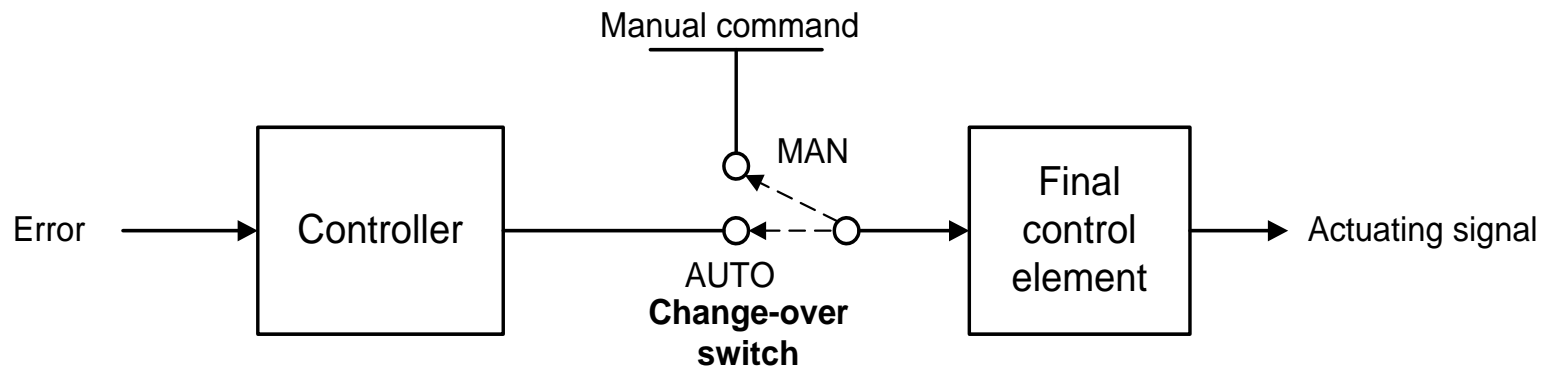
It is desirable to limit the controller output (say between 0% and 100%) so that control valves or other final control elements may operate safely within their working limits.

Provision for Auto/Manual modes of control

A change-over switch is normally provided for configuring the controller as an automatic controller (AUTO) for closed loop operation or as a manual controller (MAN) for open-loop operation.

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Provision for Local/Remote modes of operation

A provision is made for the set point input, such that, the set point may be changed either locally or from a remote link.

Electronic Process Controllers

Advantages

- **No transmission lag (as compared to pneumatic systems)**

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- **Compatibility with other electrical components**

Electronic Process Controllers

Disadvantages

- **Prone to electrical hazards (in inflammable atmosphere)**

Electronic Process Controllers

Disadvantages

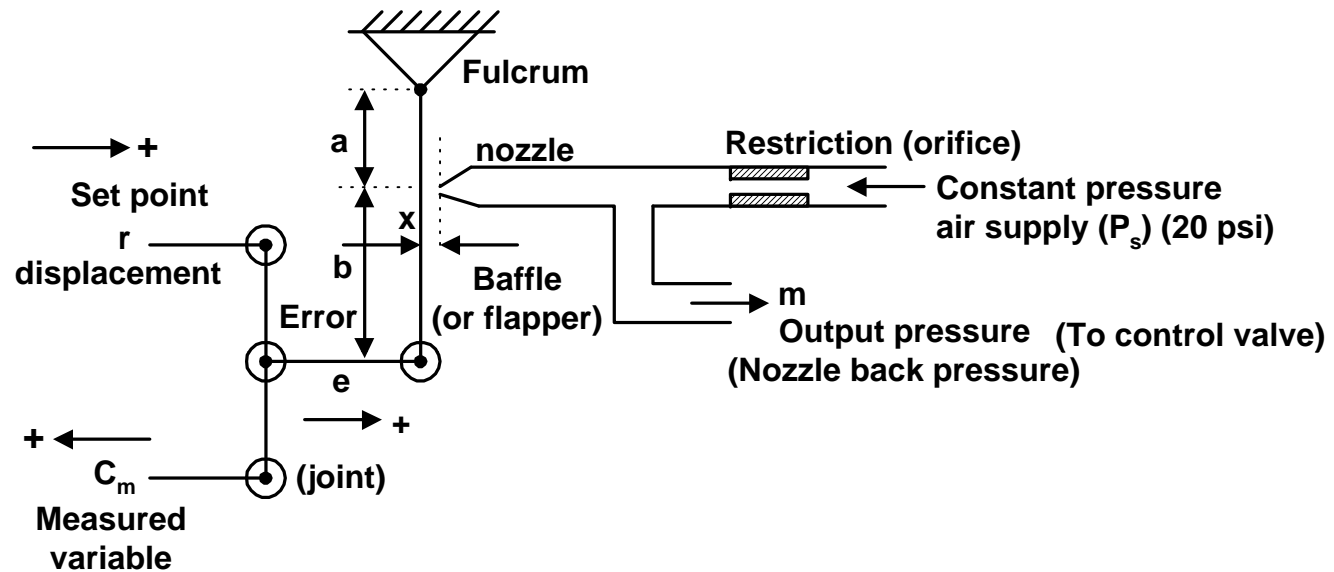
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Electronic Process Controllers

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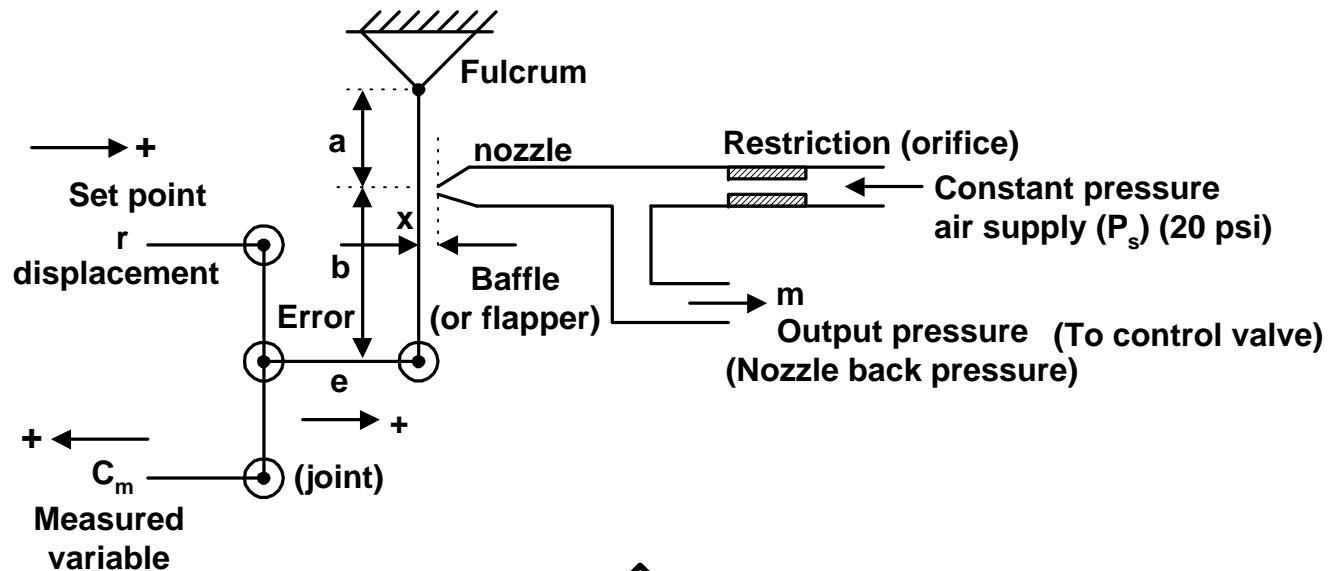
- **Prone to electrical hazards (in inflammable atmosphere)**
- **Analog integrators are not very reliable**
- **Conversion equipments are necessary to interface pneumatic and hydraulic devices**

Pneumatic Baffle-Nozzle or Flapper-Nozzle amplifier

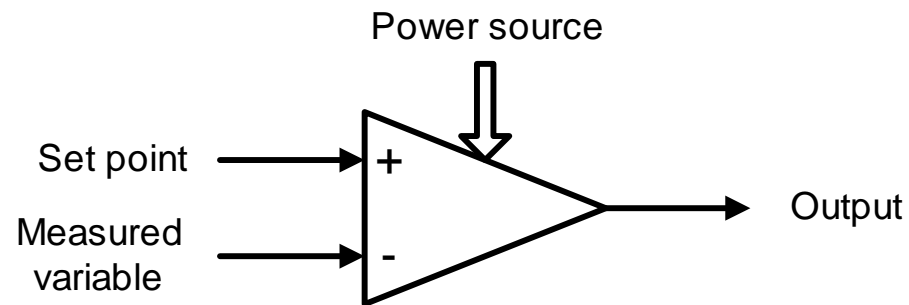
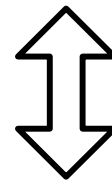


$$e = (r - C_m)/2$$

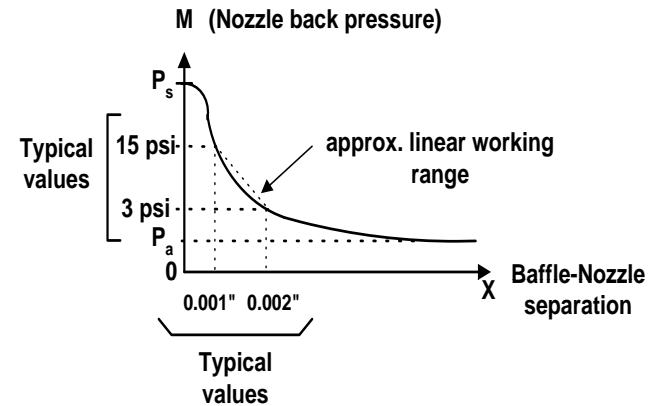
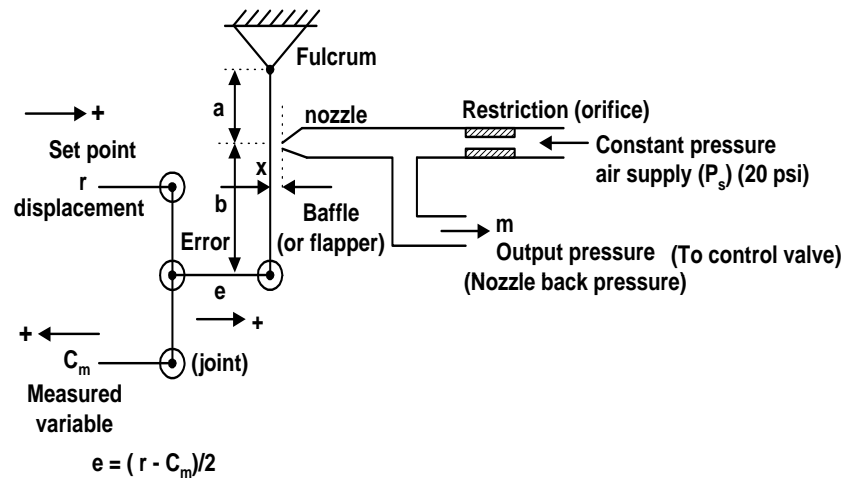
Pneumatic Baffle-Nozzle or Flapper-Nozzle amplifier



$$e = (r - C_m)/2$$



Pneumatic Baffle-Nozzle or Flapper-Nozzle amplifier



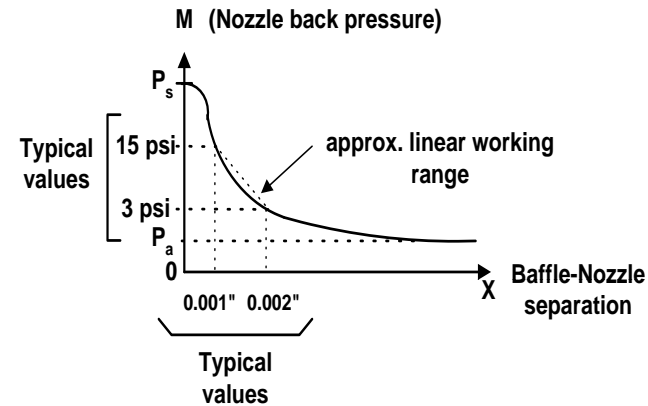
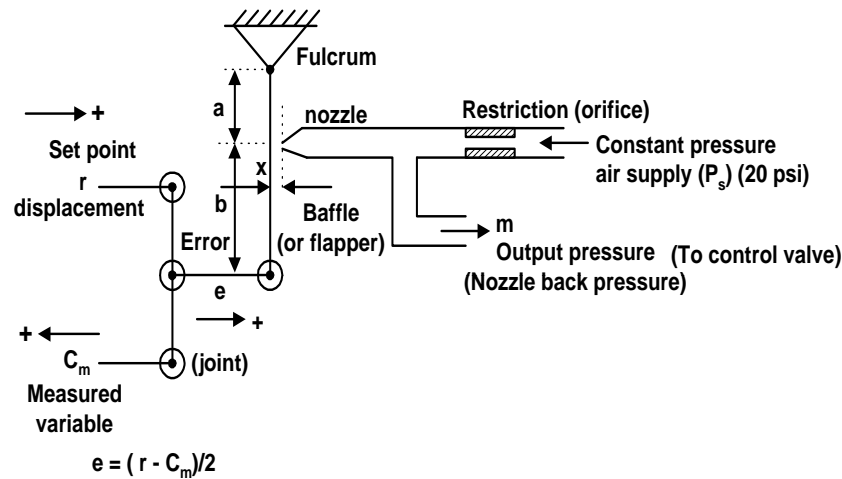
A typical curve relating nozzle back pressure M to Baffle-Nozzle separation X

x : change in baffle-nozzle separation.

m : change in back pressure.

P_a : the lowest possible pressure (= ambient pressure).

Pneumatic Baffle-Nozzle or Flapper-Nozzle amplifier



A typical curve relating nozzle back pressure M to Baffle-Nozzle separation X

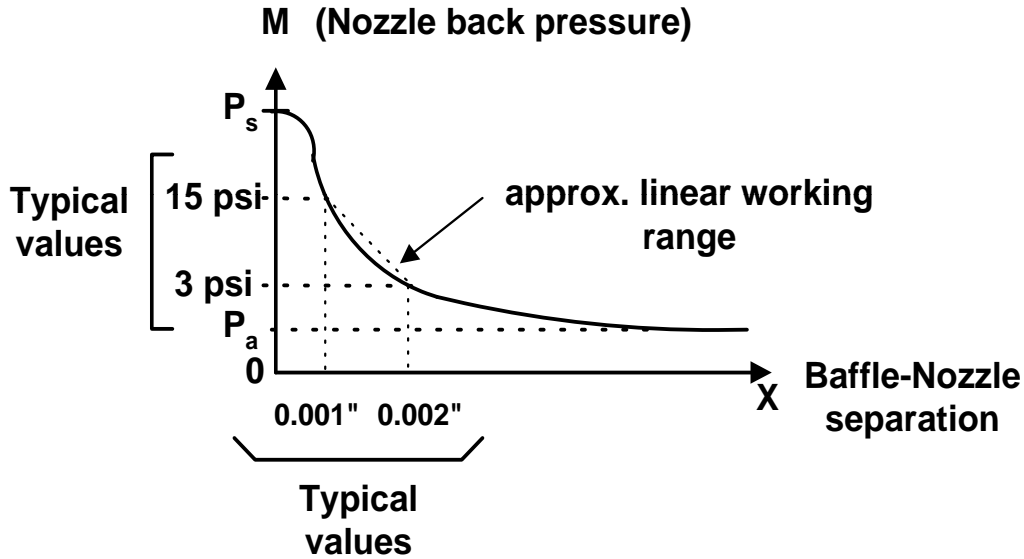
x : change in baffle-nozzle separation.

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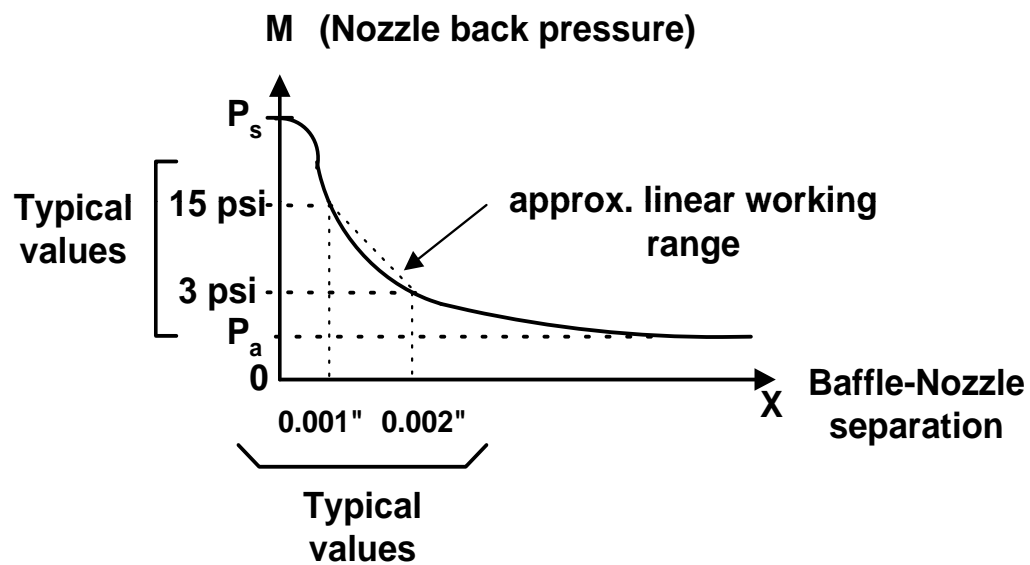
$X = X' - x$ and $M = M' + m$, X' : baffle-nozzle separation with zero error and M' : output pressure with zero error.

Pneumatic Baffle-Nozzle or Flapper-Nozzle amplifier



A typical curve relating nozzle back pressure M to Baffle-Nozzle separation X

Pneumatic Baffle-Nozzle or Flapper-Nozzle amplifier

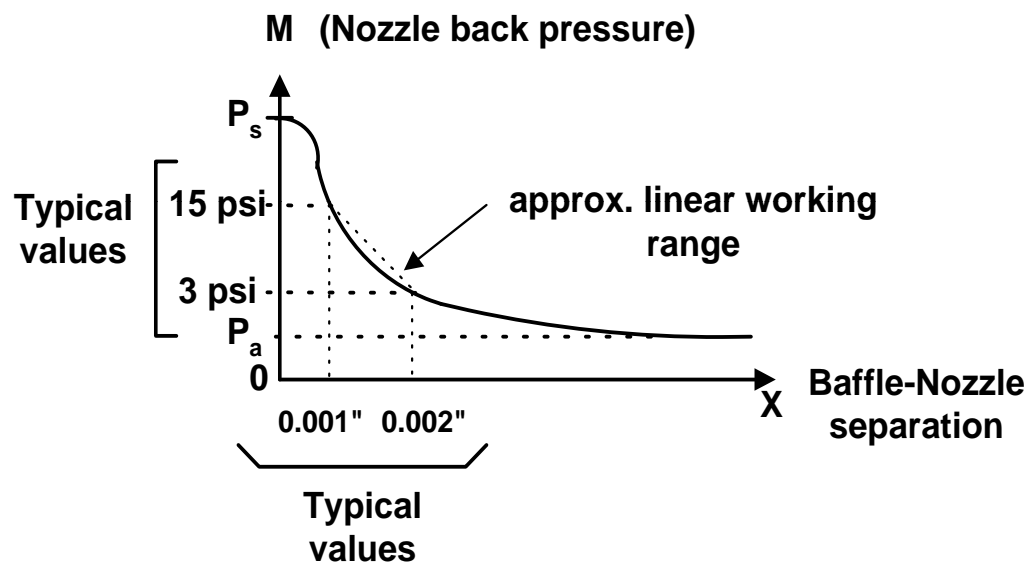


A typical curve relating nozzle back pressure M to Baffle-Nozzle separation X

For approximately linear working range:

$$M = K_n X + C \text{ and } \dot{M} = K_n \dot{X} + C, K_n \text{ being the slope}$$

Pneumatic Baffle-Nozzle or Flapper-Nozzle amplifier



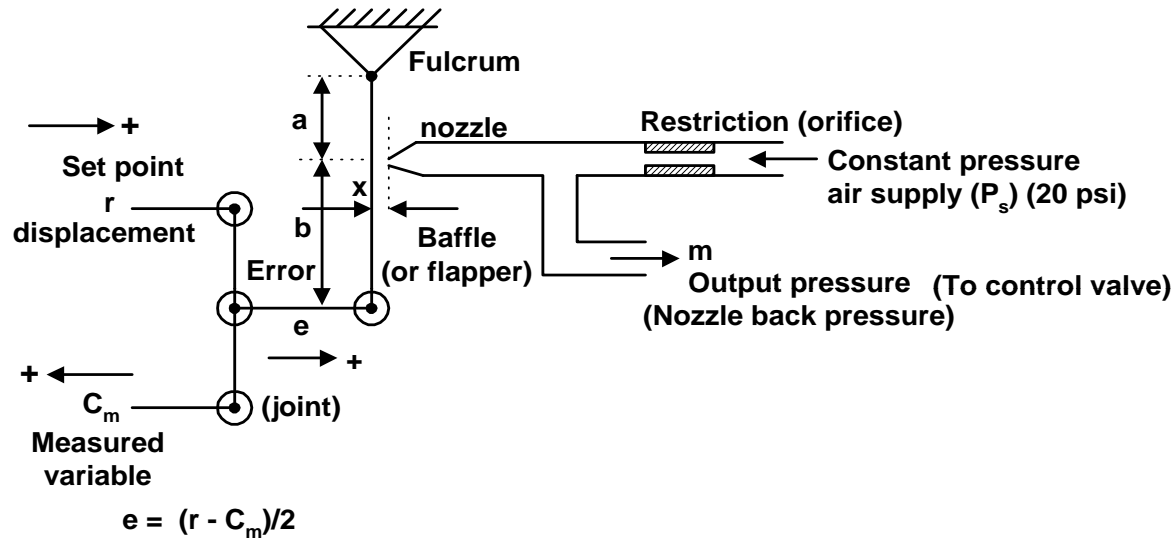
A typical curve relating nozzle back pressure M to Baffle-Nozzle separation X

For approximately linear working range:

$M = K_n X + C$ and $M' = K_n X' + C$, K_n being the slope

Subtracting, $M - M' = K_n(X - X')$, i.e., $m = -K_n \cdot x$

Pneumatic Baffle-Nozzle or Flapper-Nozzle amplifier

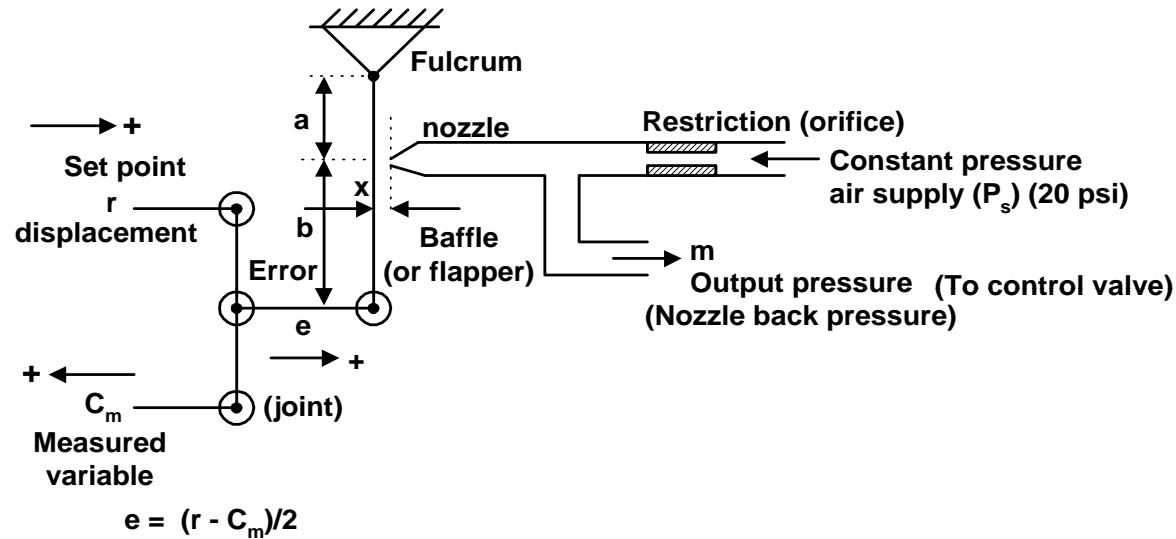


Under steady state condition, change in output pressure may be expressed as:

$$m = -K_n \cdot x, \text{ where } -K_n \text{ is the nozzle gain}$$

Here, $-x = \left(\frac{a}{a+b} \right) e$

Pneumatic Baffle-Nozzle or Flapper-Nozzle amplifier



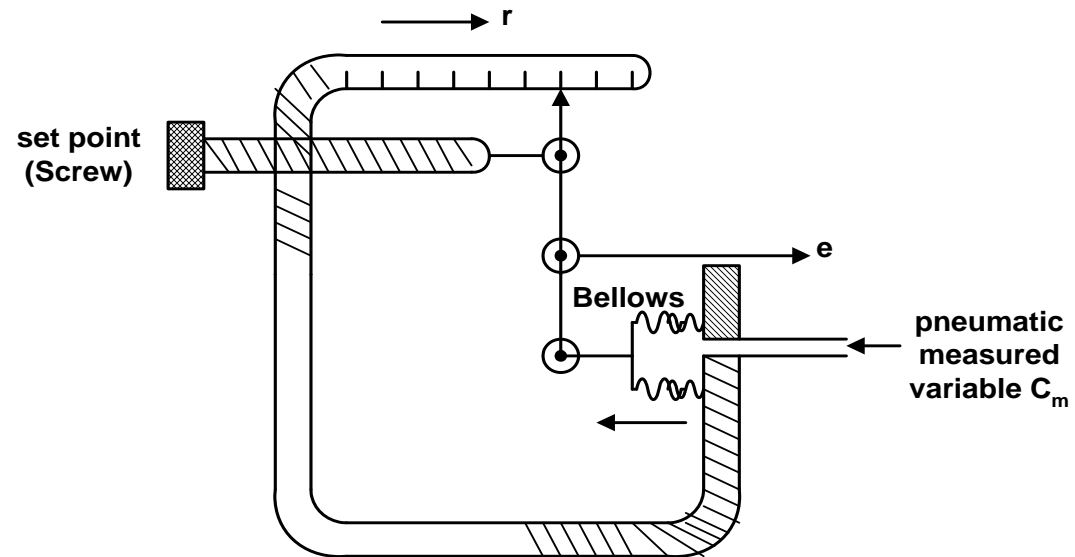
Under steady state condition, change in output pressure may be expressed as:

$$m = -K_n \cdot x, \text{ where } -K_n \text{ is the nozzle gain}$$

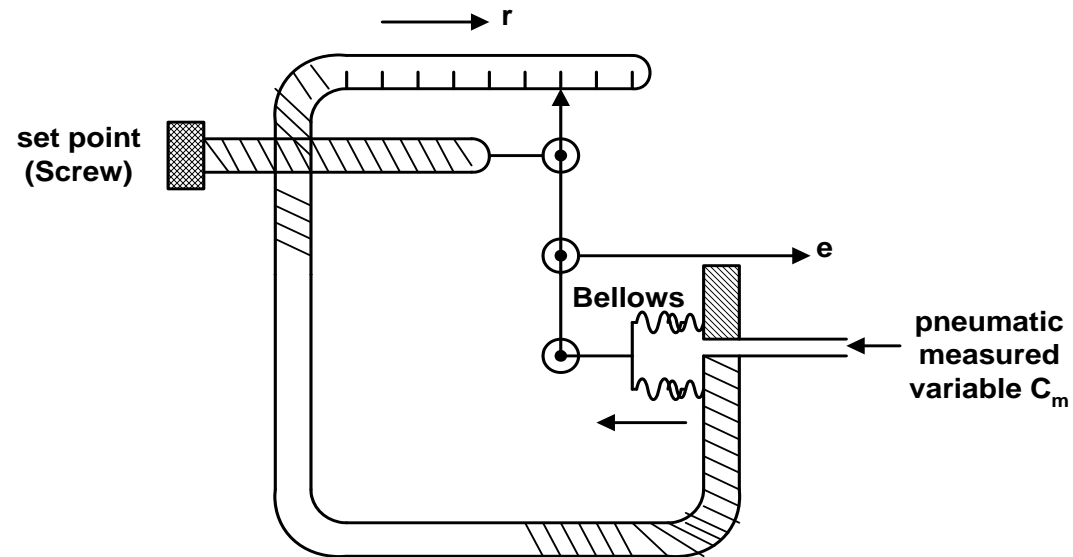
$$\text{Here, } -x = \left(\frac{a}{a+b} \right) e$$

$$\text{Thus, } m = K_n \cdot e \left(\frac{a}{a+b} \right) = K \cdot e, \text{ } K \text{ is called the amplifier gain and } K = K_n \left(\frac{a}{a+b} \right)$$

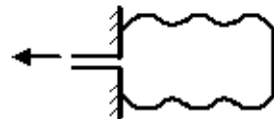
Arrangement for mechanical set-point and pneumatic measured variable



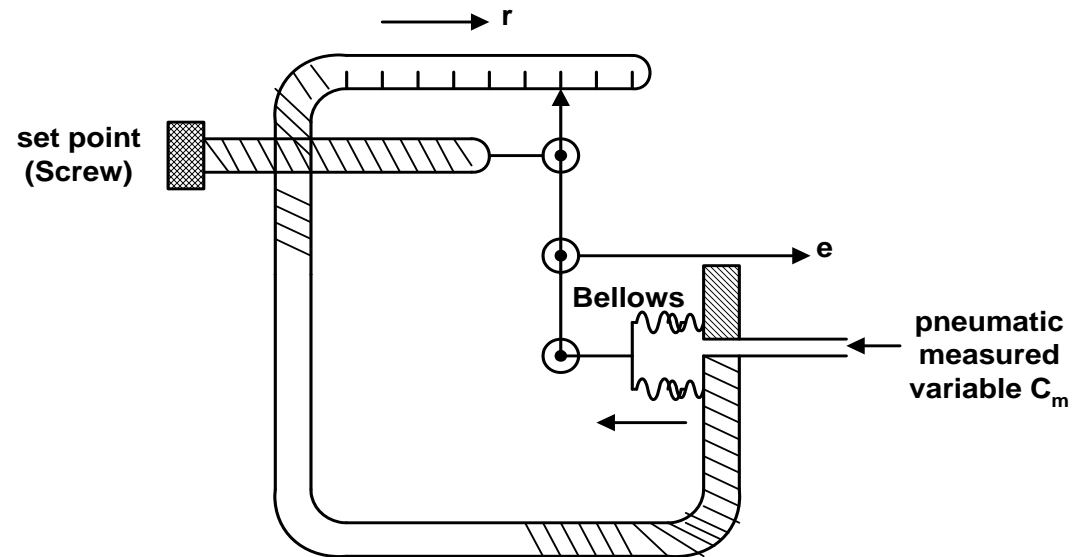
Arrangement for mechanical set-point and pneumatic measured variable



Bellows



Arrangement for mechanical set-point and pneumatic measured variable



Bellows



Relay valve or pilot valve

The output pressure from the pneumatic amplifier is not suitable for driving the final control element due to the presence of restriction in the air supply.

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For this reason, a buffer stage, known as *relay valve* or *pilot valve*, is added at the output to allow sufficient air flow at pressure 'm'.

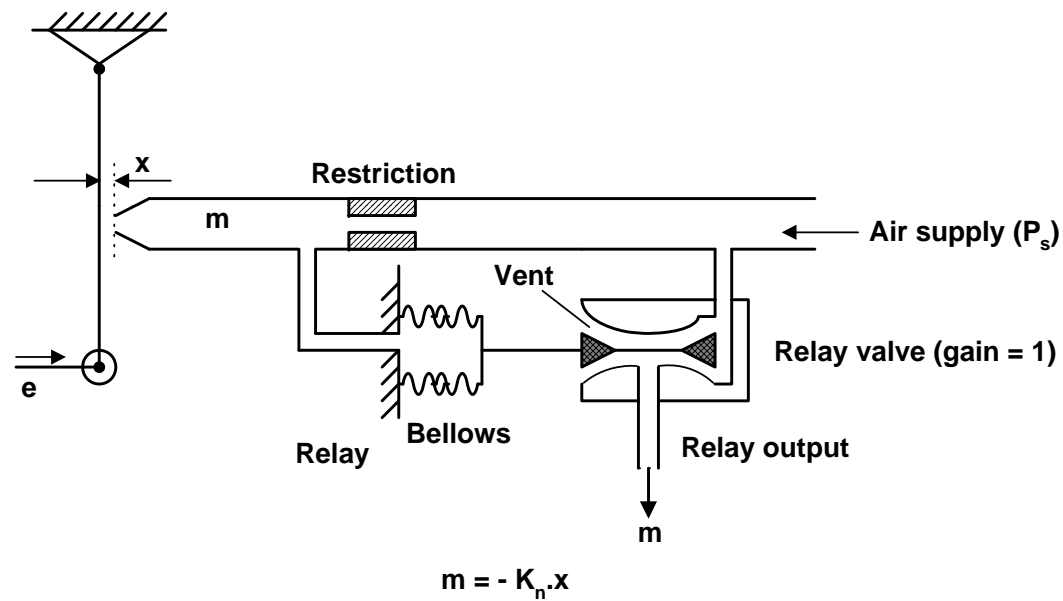
Relay valve or pilot valve

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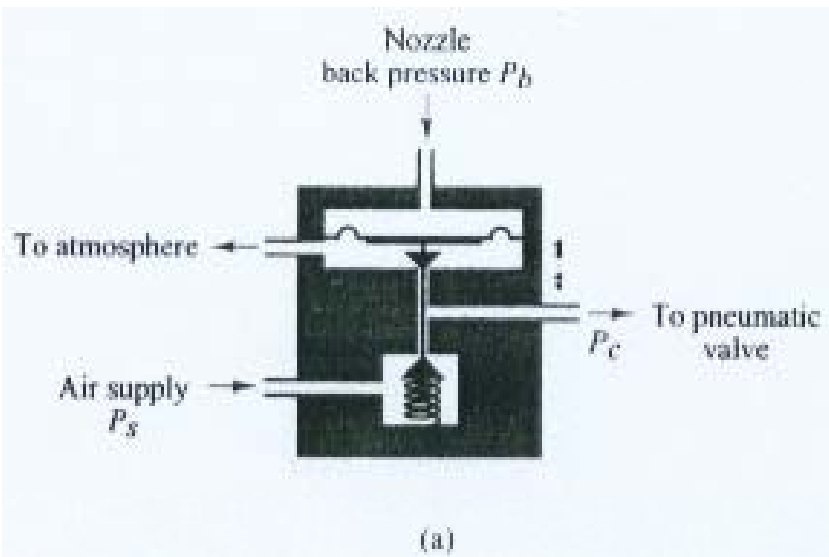
A relay valve may be *direct-acting* (positive gain) or *reverse-acting* (negative gain).

Direct-acting relay valve with a Baffle-Nozzle amplifier



As the nozzle back pressure m increases, the relay output pressure also increases

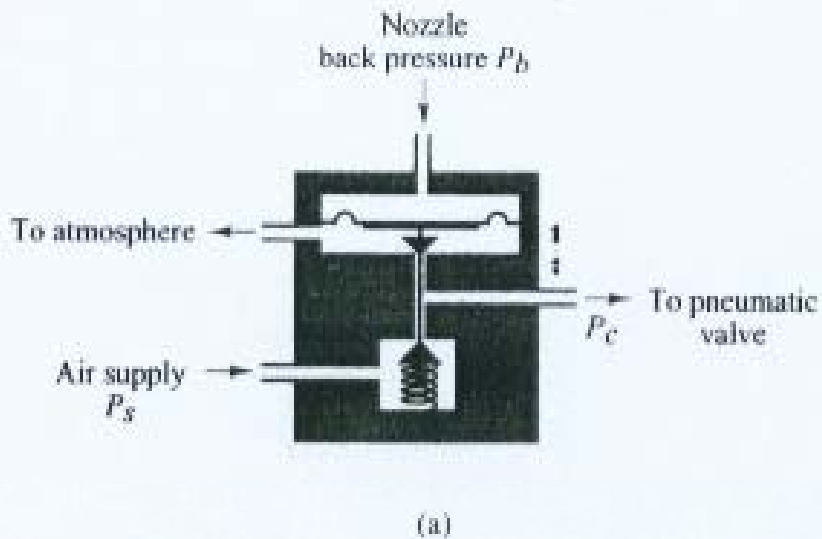
Direct-acting relays



Bleed type relay

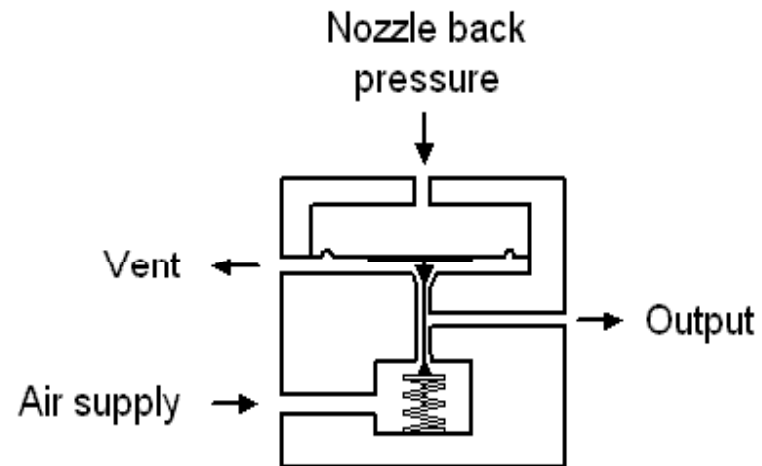
In all positions of the valve, except at the position to shut off the air supply, air continues to bleed into the atmosphere.

Direct-acting relays

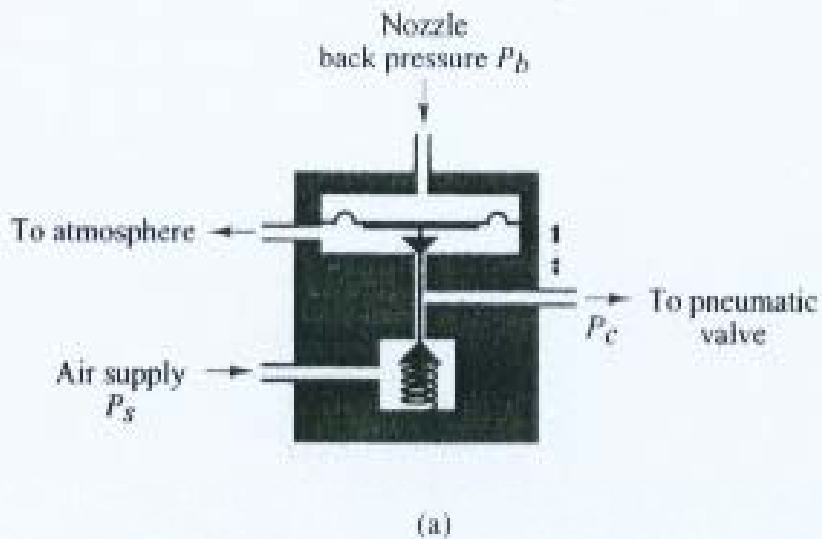


Bleed type relay

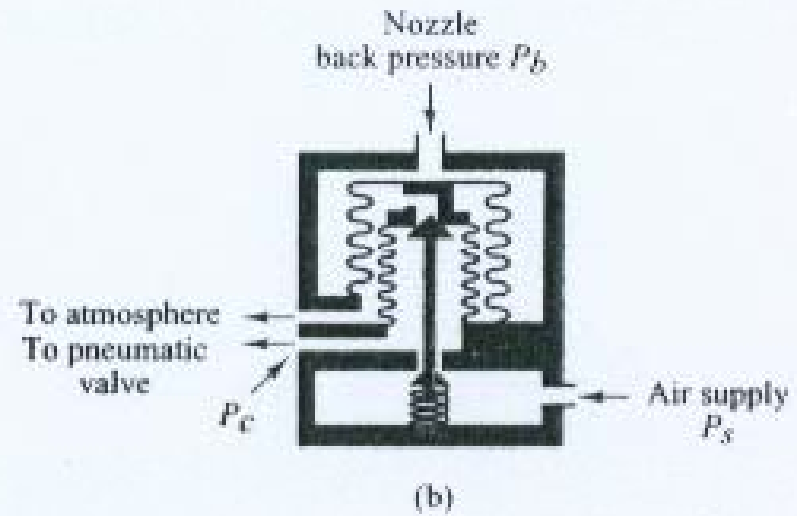
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Direct-acting relays



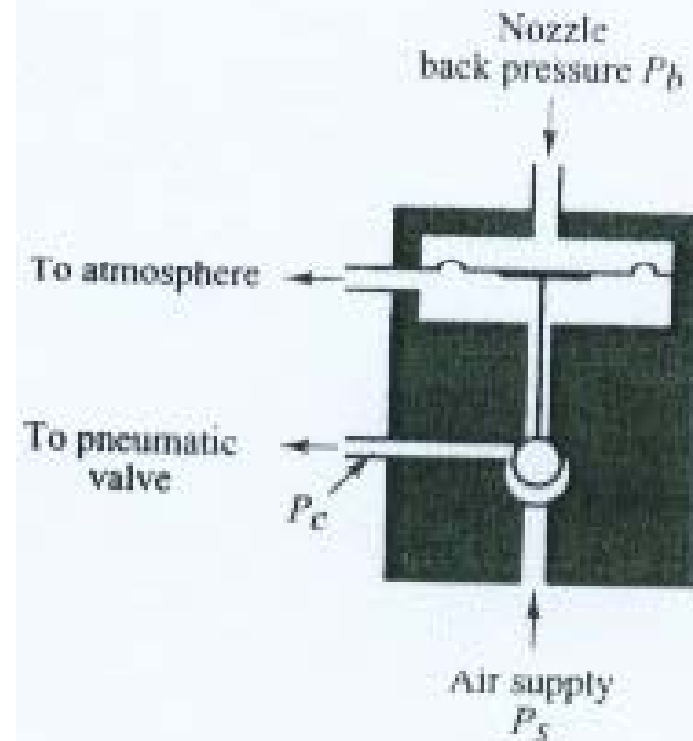
Bleed type relay



Non-bleed type relay

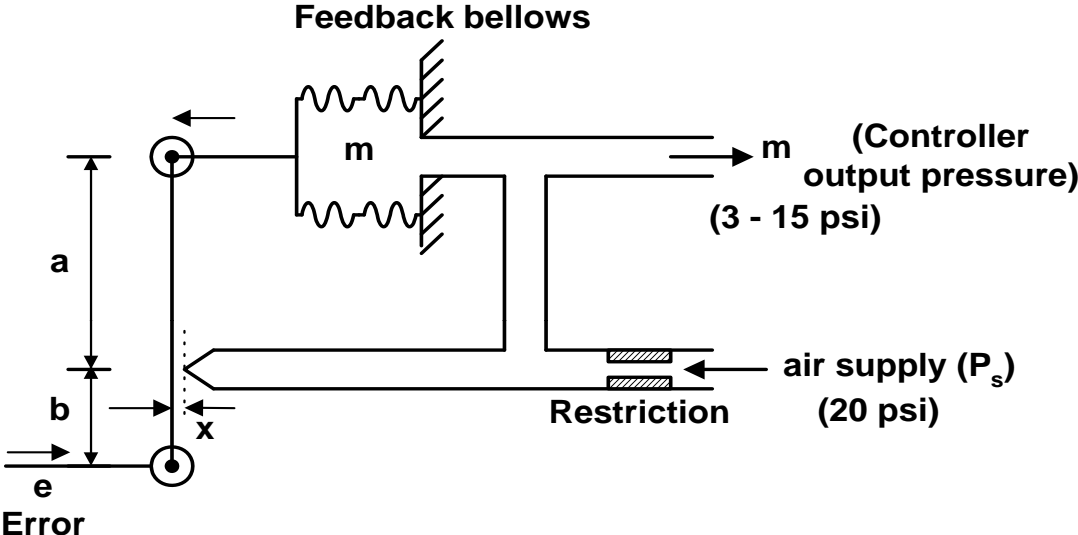
The air bleed stops when the equilibrium condition is obtained, and, therefore, there is no loss of pressurized air at steady-state operation.

Reverse-acting relay

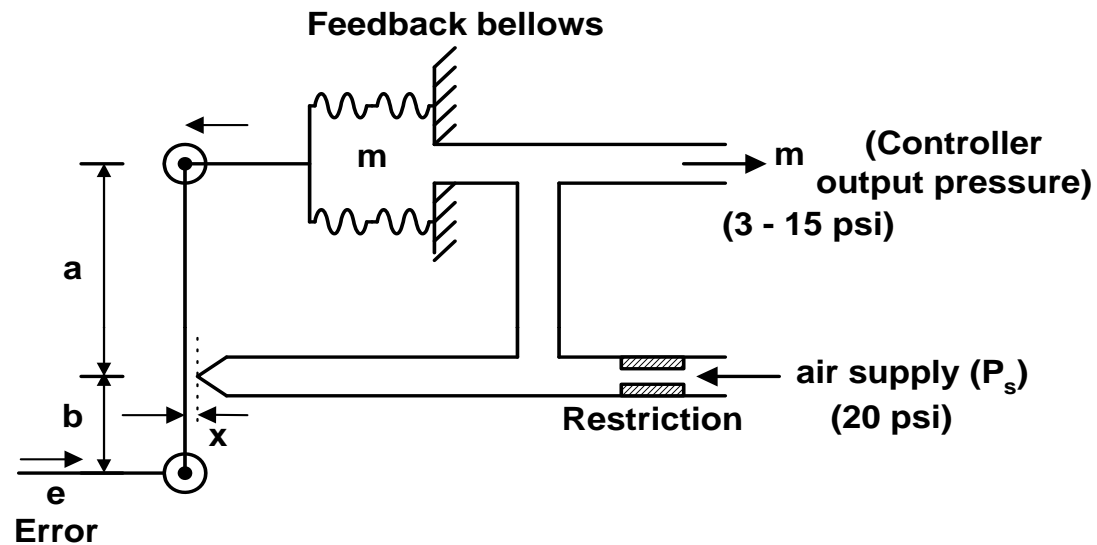


As the nozzle back pressure increases, the ball valve is forced towards the lower seat, thereby decreasing the output pressure

Pneumatic Proportional Controller



Pneumatic Proportional Controller

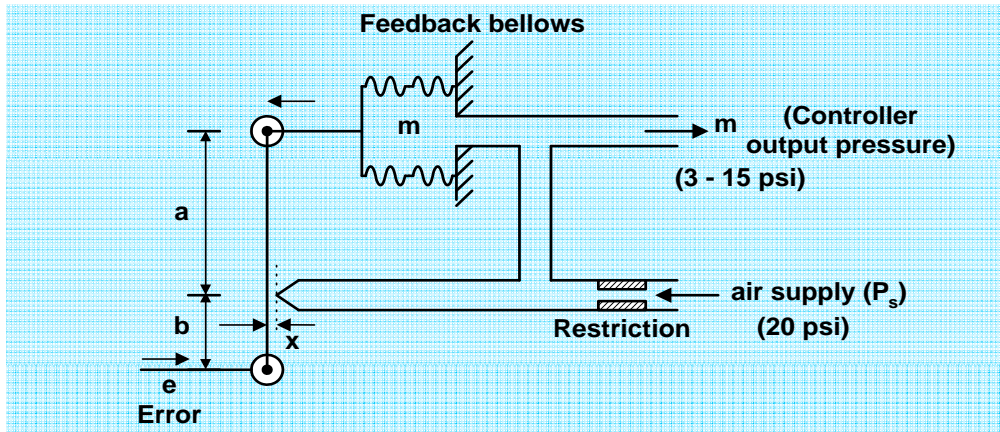


The Baffle-Nozzle separation may be expressed as:

$$-(x) = \left(\frac{a}{a+b} \right) e - \left(\frac{b}{a+b} \right) K_b m$$

where K_b is the bellows stiffness factor and m is the change in output pressure.

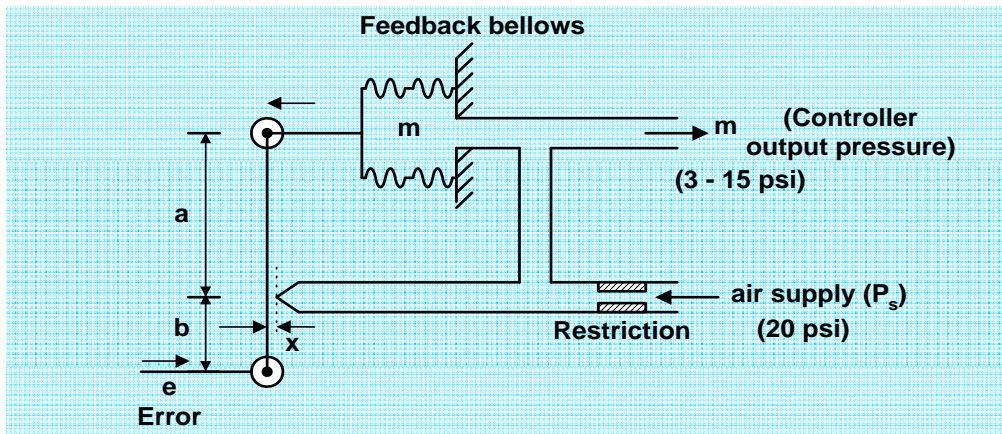
Pneumatic Proportional Controller



$$-(x) = \left(\frac{a}{a+b} \right) e - \left(\frac{b}{a+b} \right) K_b m$$

Now $m = -K_n .x$, where K_n is the nozzle gain.

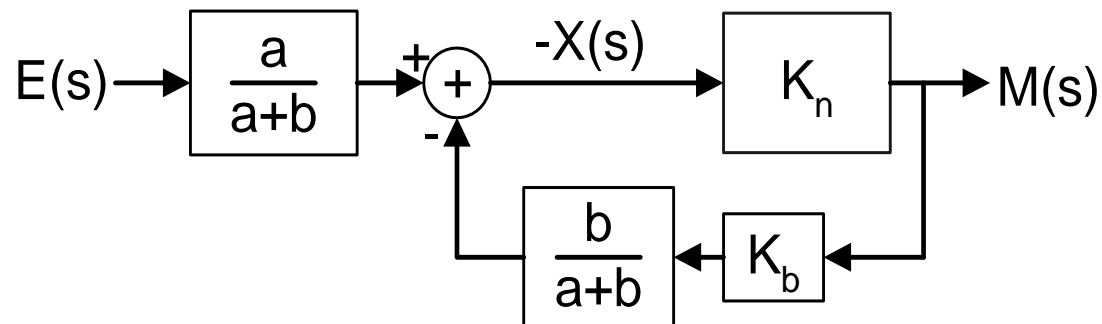
Pneumatic Proportional Controller



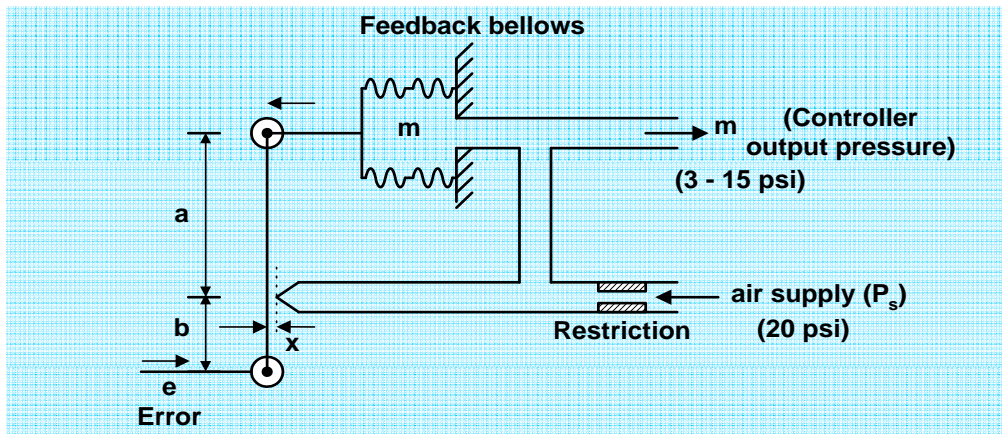
$$-(x) = \left(\frac{a}{a+b} \right) e - \left(\frac{b}{a+b} \right) K_b m$$

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Block diagram of the controller



Pneumatic Proportional Controller



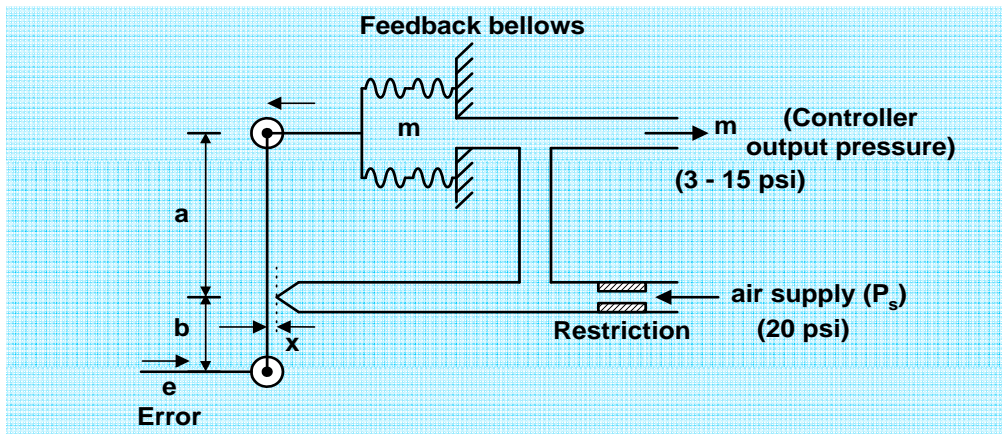
$$-(x) = \left(\frac{a}{a+b} \right) e - \left(\frac{b}{a+b} \right) K_b m$$

Now $m = -K_n .x$, where K_n is the nozzle gain.

Thus,

$$\frac{m}{K_n} = \left(\frac{a}{a+b} \right) e - \left(\frac{b}{a+b} \right) K_b m$$

Pneumatic Proportional Controller



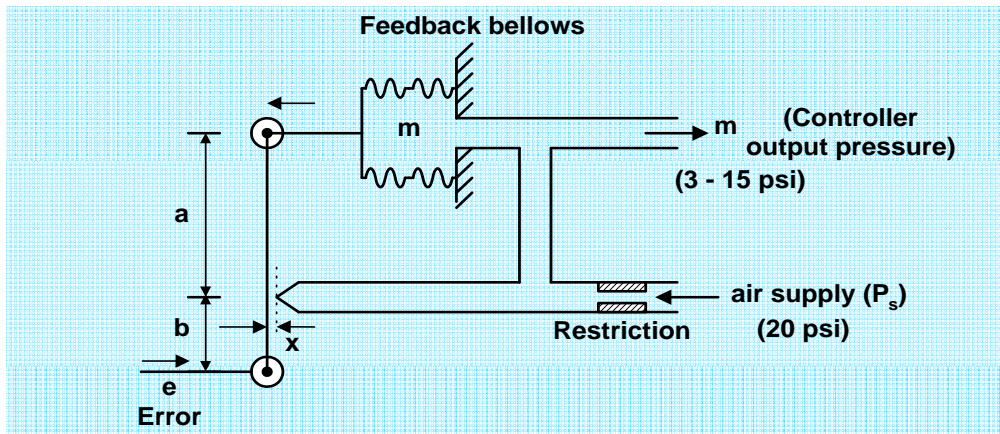
$$-x = \left(\frac{a}{a+b} \right) e - \left(\frac{b}{a+b} \right) K_b m$$

Now $m = -K_n \cdot x$, where K_n is the nozzle gain.

Thus,
$$\frac{m}{K_n} = \left(\frac{a}{a+b} \right) e - \left(\frac{b}{a+b} \right) K_b m$$

or
$$m \left[\frac{1}{K_n} + \left(\frac{b}{a+b} \right) K_b \right] = \left(\frac{a}{a+b} \right) e$$

Pneumatic Proportional Controller



$$-(x) = \left(\frac{a}{a+b} \right) e - \left(\frac{b}{a+b} \right) K_b m$$

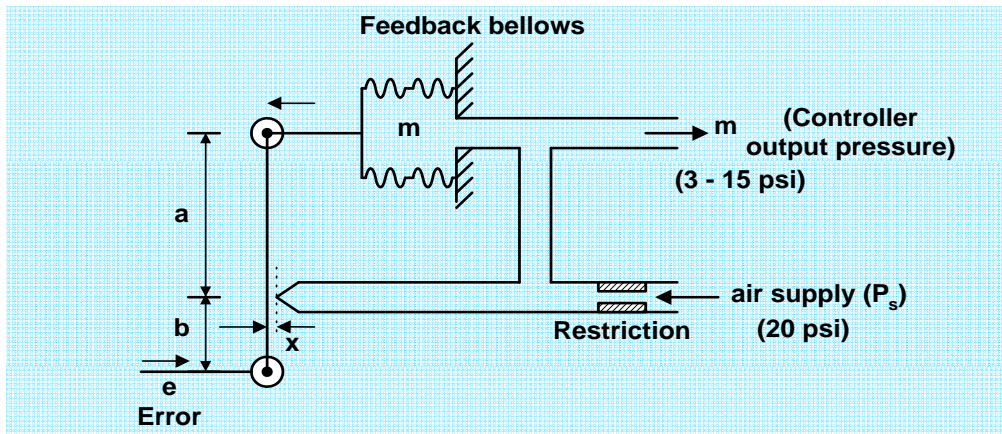
Now $m = -K_n \cdot x$, where K_n is the nozzle gain.

Thus,
$$\frac{m}{K_n} = \left(\frac{a}{a+b} \right) e - \left(\frac{b}{a+b} \right) K_b m$$

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$$m \left[\frac{1}{K_n} + \left(\frac{b}{a+b} \right) K_b \right] = \left(\frac{a}{a+b} \right) e$$

if K_n is very high, then
$$\frac{1}{K_n} \approx 0,$$

Pneumatic Proportional Controller

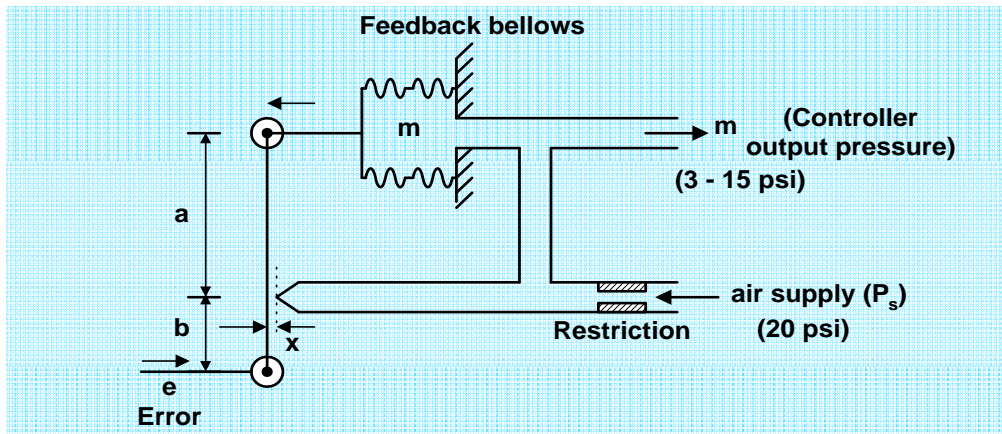


This gives

$$m = \frac{\left(\frac{a}{a+b}\right)}{\left(\frac{b}{a+b}\right) K_b} e = \left(\frac{a}{bK_b}\right) e = K_p \cdot e$$

where $K_p = \text{proportional gain} = \frac{a}{bK_b}$

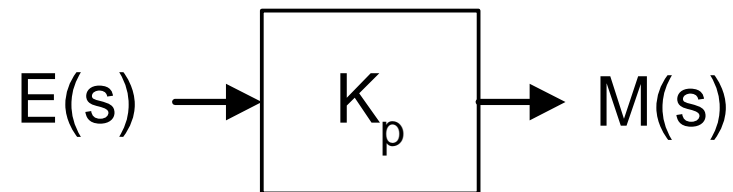
Pneumatic Proportional Controller



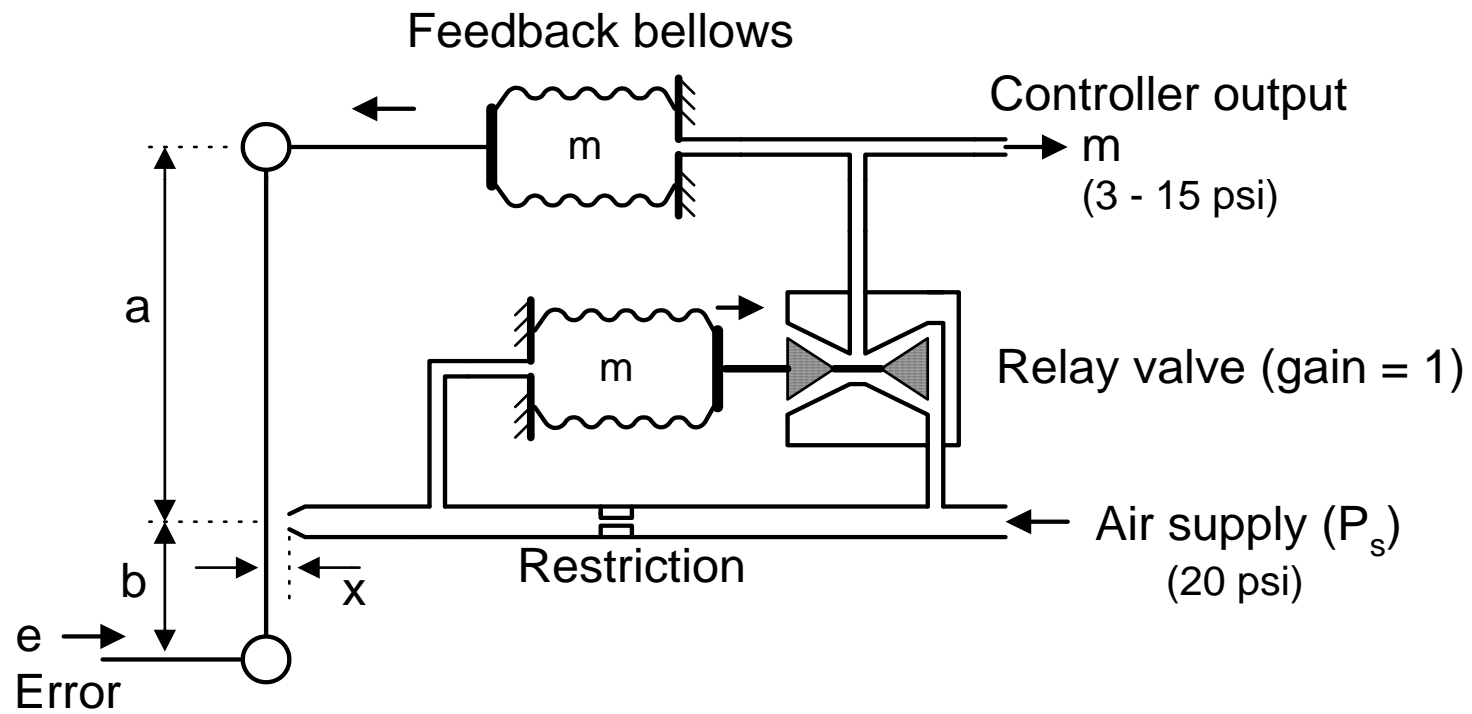
$$m = \frac{\left(\frac{a}{a+b}\right)}{\left(\frac{b}{a+b}\right) K_b} e = \left(\frac{a}{bK_b}\right) e = K_p \cdot e$$

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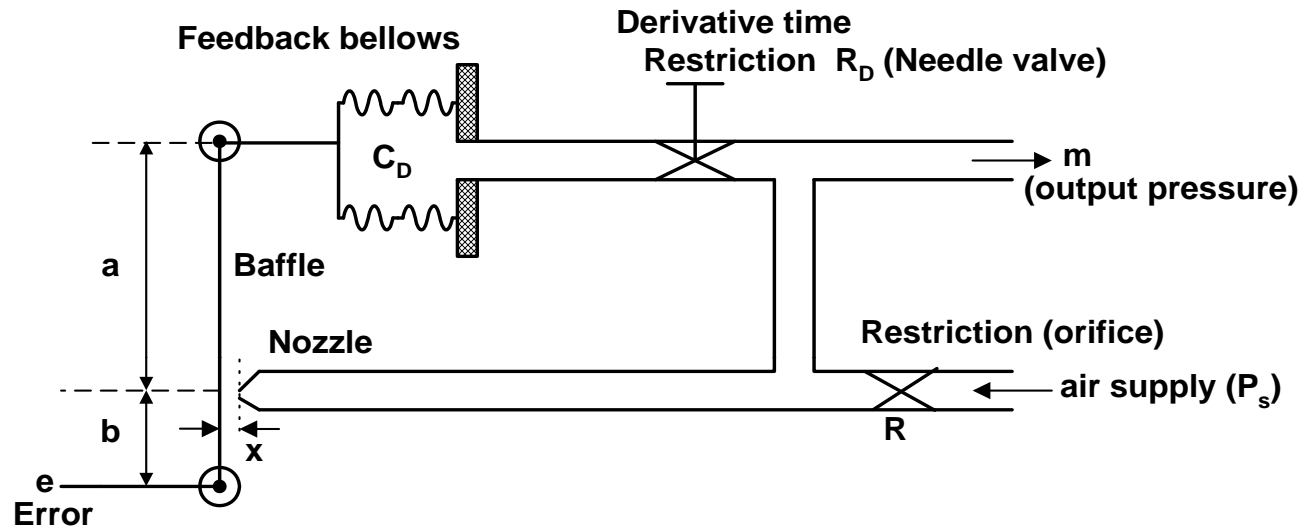
Simplified block diagram of the controller



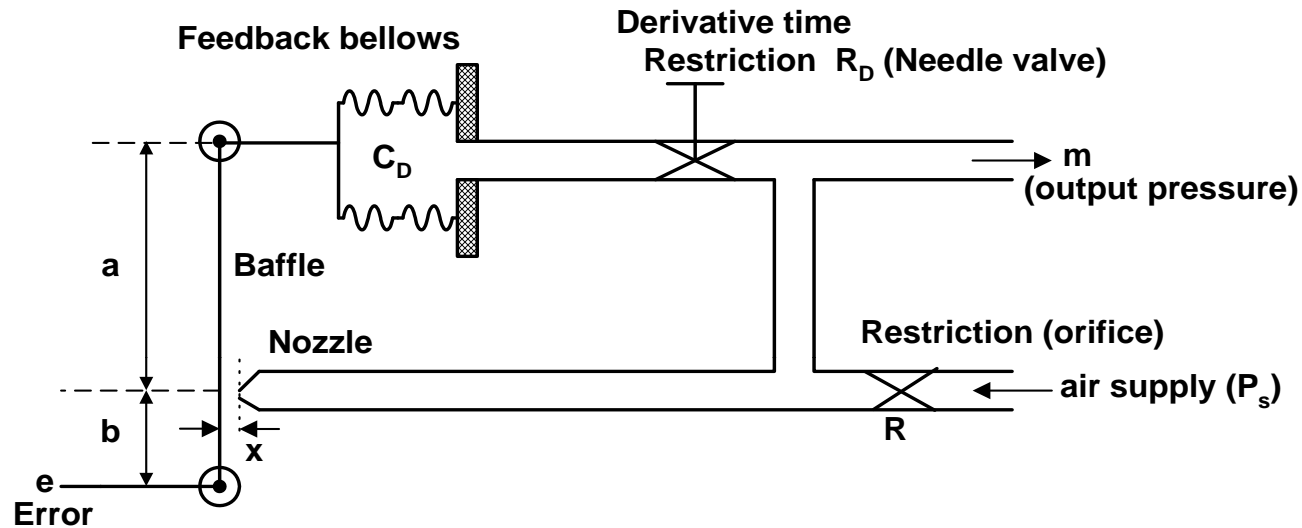
Pneumatic proportional controller with a direct-acting relay



Pneumatic Proportional-Derivative Controller



Pneumatic Proportional-Derivative Controller

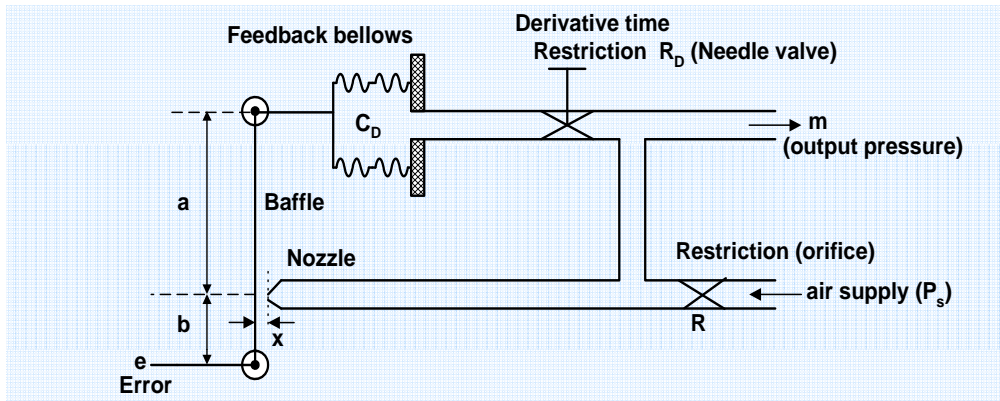


The Baffle-Nozzle separation may be expressed as:

$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s) \quad (\text{assuming } R_D \gg R)$$

where K_b = Bellows stiffness factor,
 T_D = Derivative time
 $= R_D C_D$, (assuming $R_D \gg R$),
 C_D = Capacity of the bellows.

Pneumatic Proportional-Derivative Controller

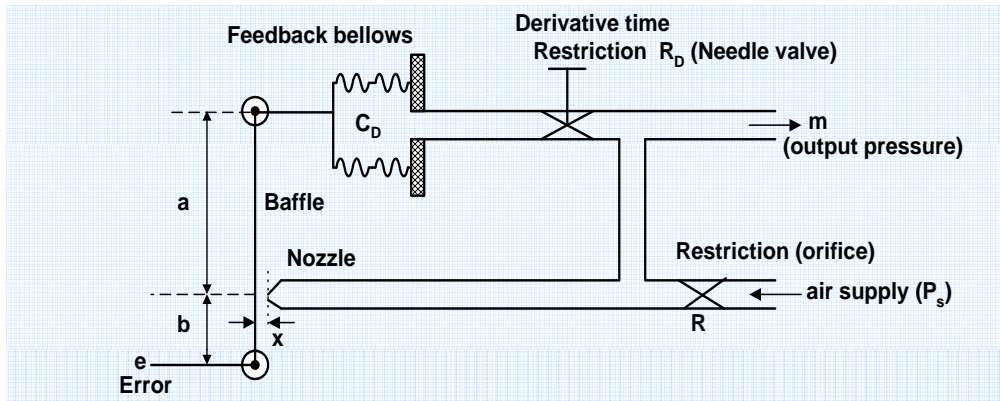


$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s)$$

$$\text{Now } M(s) = -K_n \cdot X(s)$$

where K_n is the nozzle gain.

Pneumatic Proportional-Derivative Controller



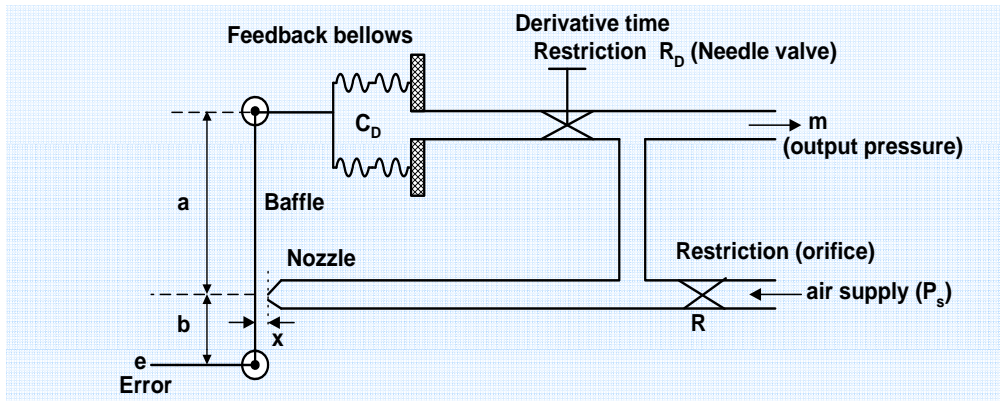
$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s)$$

$$\text{Now } M(s) = -K_n \cdot X(s)$$

where K_n is the nozzle gain.

$$\text{Thus } \frac{M(s)}{K_n} = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s)$$

Pneumatic Proportional-Derivative Controller



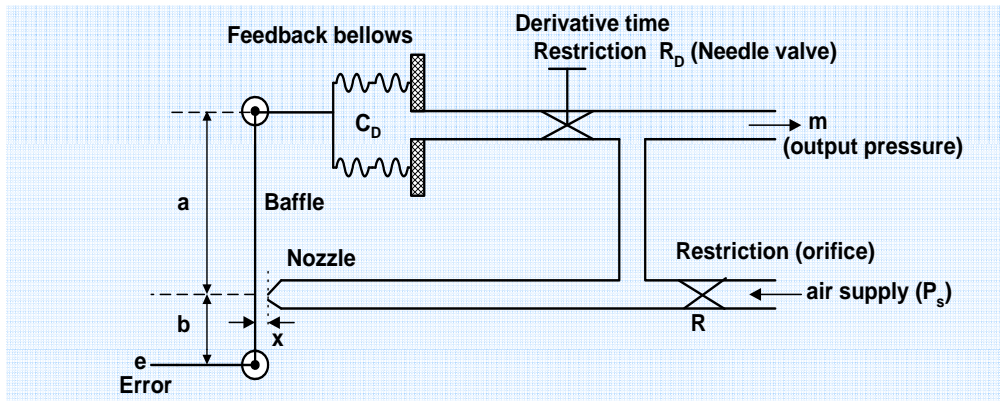
$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s)$$

Now $M(s) = -K_n \cdot X(s)$
 where K_n is the nozzle gain.

Thus
$$\frac{M(s)}{K_n} = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s)$$

or,
$$M(s) \left[\frac{1}{K_n} + \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) \right] = \left(\frac{a}{a+b} \right) E(s)$$

Pneumatic Proportional-Derivative Controller



$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s)$$

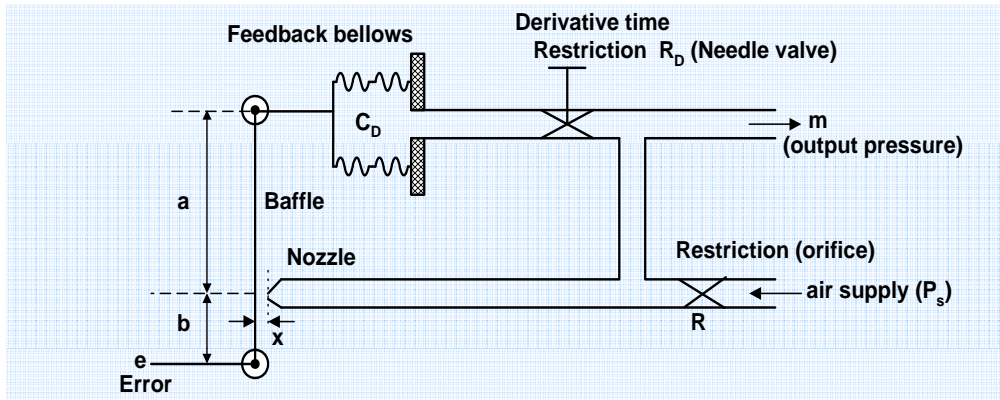
$$\text{Now } M(s) = -K_n \cdot X(s)$$

where K_n is the nozzle gain.

$$\text{Thus } \frac{M(s)}{K_n} = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s)$$

$$\text{or, } M(s) \left[\frac{1}{K_n} + \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) \right] = \left(\frac{a}{a+b} \right) E(s) \quad \text{Now, } K_n \gg 1, \frac{1}{K_n} \approx 0,$$

Pneumatic Proportional-Derivative Controller



$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s)$$

Now $M(s) = -K_n \cdot X(s)$
where K_n is the nozzle gain.

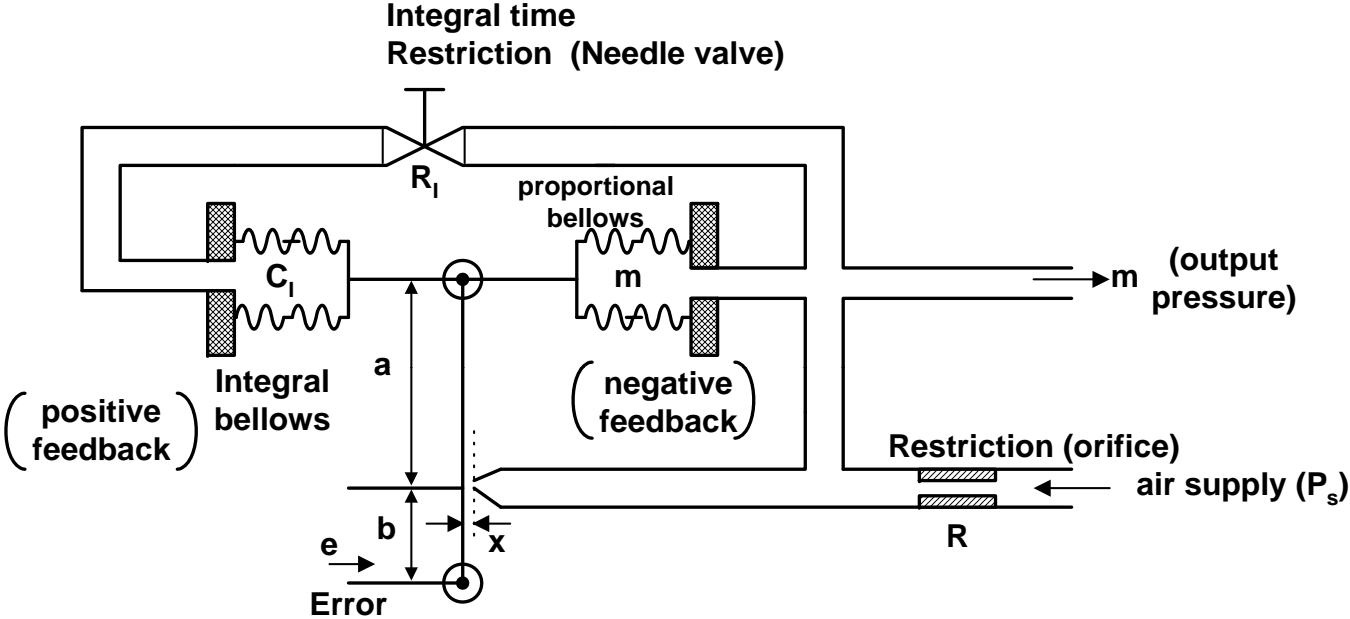
$$\text{Thus } \frac{M(s)}{K_n} = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s)$$

$$\text{or, } M(s) \left[\frac{1}{K_n} + \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) \right] = \left(\frac{a}{a+b} \right) E(s) \quad \text{Now, } K_n \gg 1, \frac{1}{K_n} \approx 0,$$

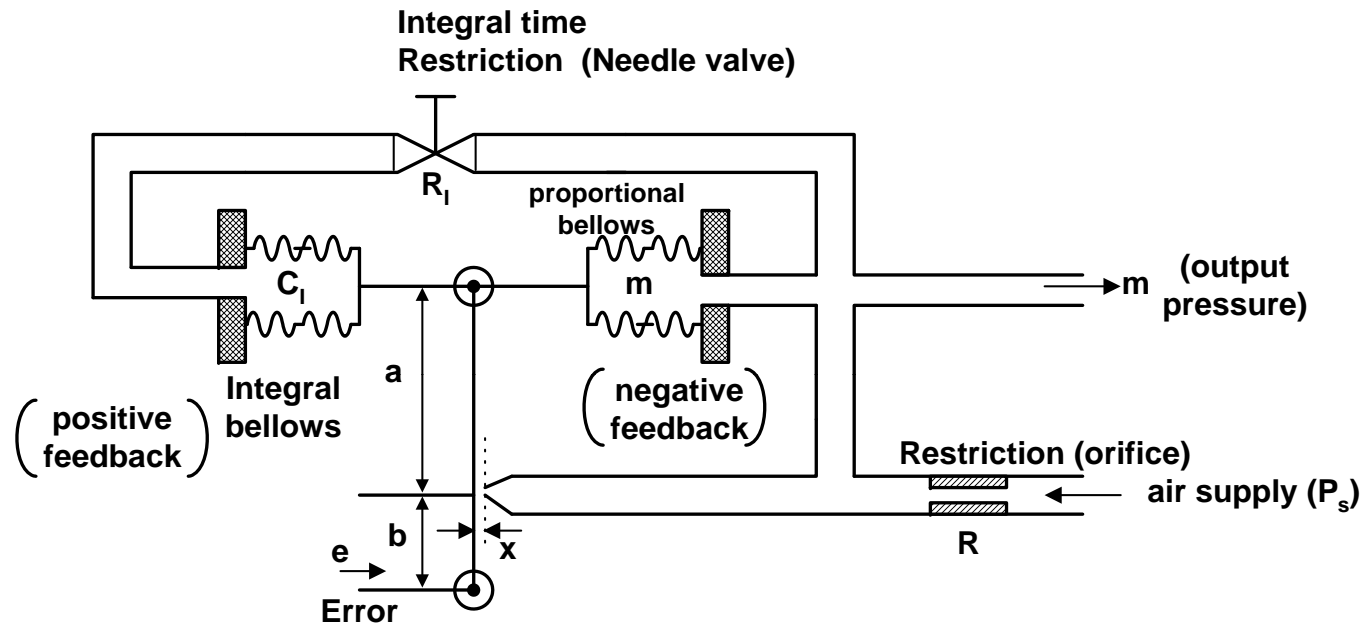
$$\therefore \frac{M(s)}{E(s)} \approx \left(\frac{a}{bK_b} \right) (1 + sT_D)$$

$$= K_p (1 + sT_D) \quad \text{where } K_p = \text{proportional gain} = \frac{a}{bK_b}$$

Pneumatic Proportional-Integral Controller



Pneumatic Proportional-Integral Controller



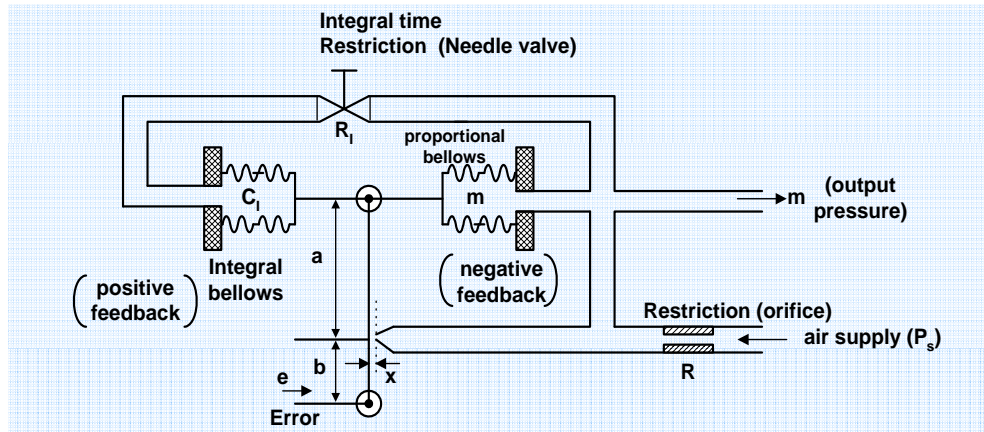
The Baffle-Nozzle separation may be expressed as: (assuming $R_1 \gg R$ and same stiffness for both bellows)

$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) K_b M(s) + \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_i} \right) M(s)$$

where $T_i =$ Integral time
 $= R_1 C_1$

and $C_1 =$ Capacity of integral bellows
 $K_b =$ Bellows stiffness factor.

Pneumatic Proportional-Integral Controller

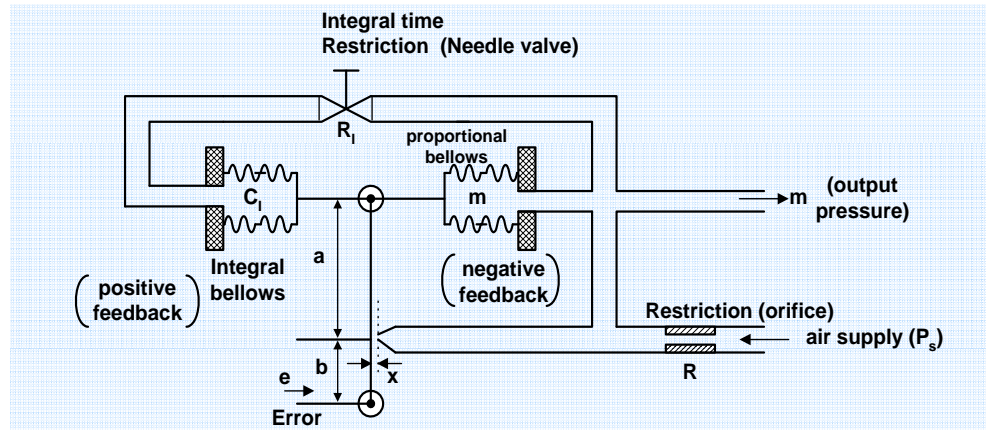


$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) K_b M(s) + \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_i} \right) M(s)$$

Hence,

$$\begin{aligned} -X(s) &= \left(\frac{a}{a+b} \right) E(s) - \left(1 - \frac{1}{1+sT_i} \right) \left(\frac{b}{a+b} \right) K_b M(s) \\ &= \left(\frac{a}{a+b} \right) E(s) - \left(\frac{1}{1 + \frac{1}{sT_i}} \right) \left(\frac{b}{a+b} \right) K_b M(s) \end{aligned}$$

Pneumatic Proportional-Integral Controller



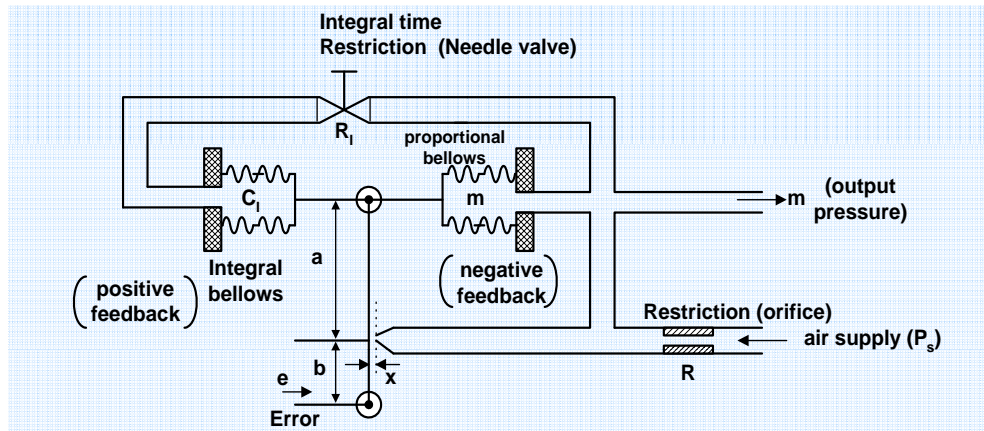
$$-X(s) = \left(\frac{a}{a+b}\right)E(s) - \left(\frac{b}{a+b}\right)K_b M(s) + \left(\frac{b}{a+b}\right)\left(\frac{K_b}{1+sT_i}\right)M(s)$$

Hence,

$$\begin{aligned} -X(s) &= \left(\frac{a}{a+b}\right)E(s) - \left(1 - \frac{1}{1+sT_i}\right)\left(\frac{b}{a+b}\right)K_b M(s) \\ &= \left(\frac{a}{a+b}\right)E(s) - \left(\frac{1}{1+\frac{1}{sT_i}}\right)\left(\frac{b}{a+b}\right)K_b M(s) \end{aligned}$$

Now, $M(s) = -K_n \cdot X(s)$

Pneumatic Proportional-Integral Controller



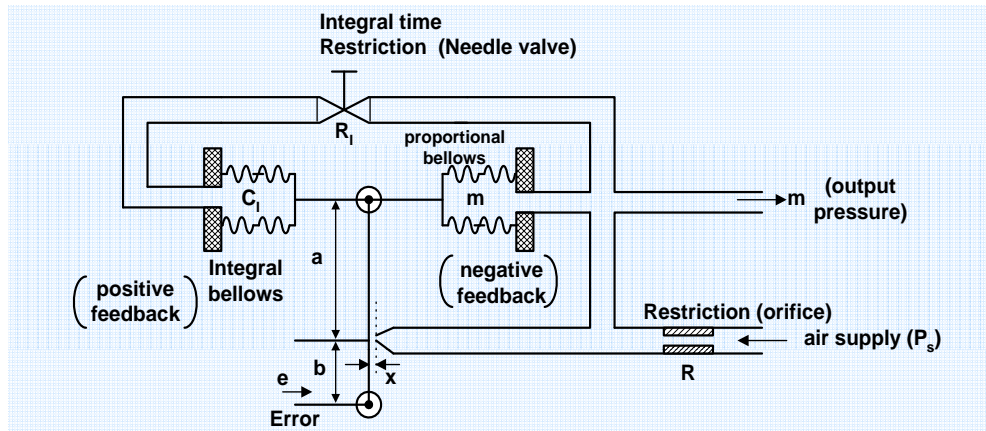
$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(1 - \frac{1}{1+sT_i} \right) \left(\frac{b}{a+b} \right) K_b M(s)$$

$$= \left(\frac{a}{a+b} \right) E(s) - \left(\frac{1}{1 + \frac{1}{sT_i}} \right) \left(\frac{b}{a+b} \right) K_b M(s)$$

and $M(s) = -K_n \cdot X(s)$

Thus, assuming $K_n \gg 1$, i.e. $\frac{1}{K_n} \approx 0$,

Pneumatic Proportional-Integral Controller



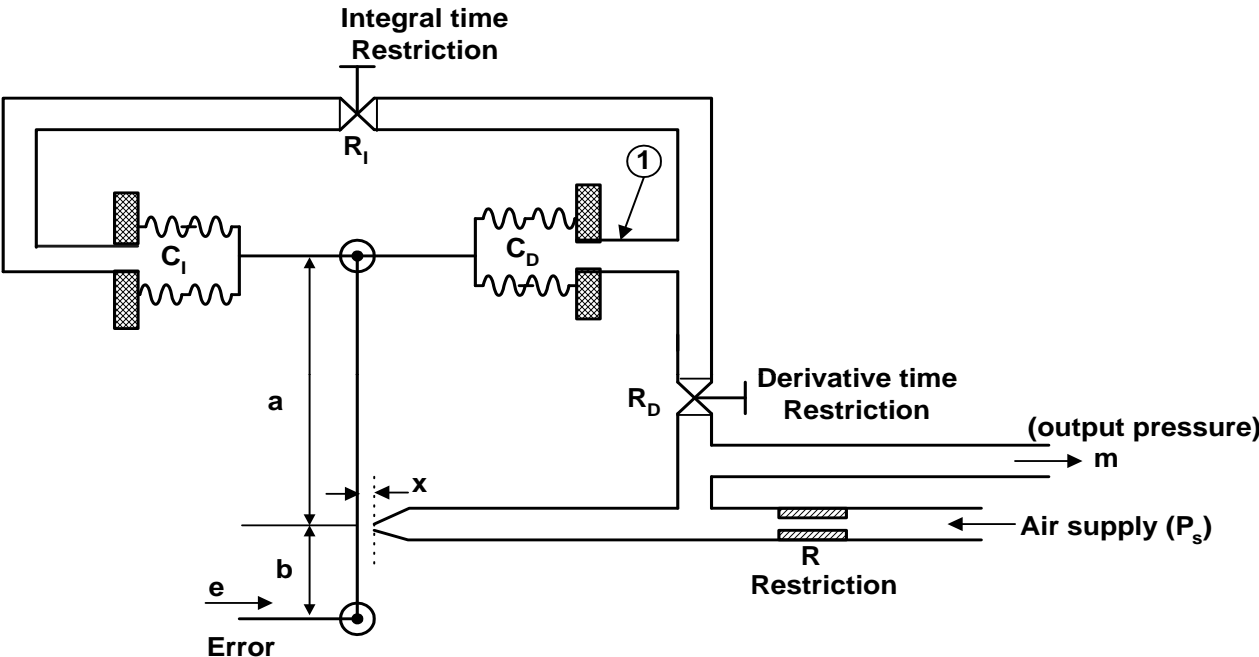
$$\begin{aligned}
 -X(s) &= \left(\frac{a}{a+b} \right) E(s) - \left(1 - \frac{1}{1+sT_i} \right) \left(\frac{b}{a+b} \right) K_b M(s) \\
 &= \left(\frac{a}{a+b} \right) E(s) - \left(\frac{1}{1+\frac{1}{sT_i}} \right) \left(\frac{b}{a+b} \right) K_b M(s)
 \end{aligned}$$

and $M(s) = -K_n \cdot X(s)$

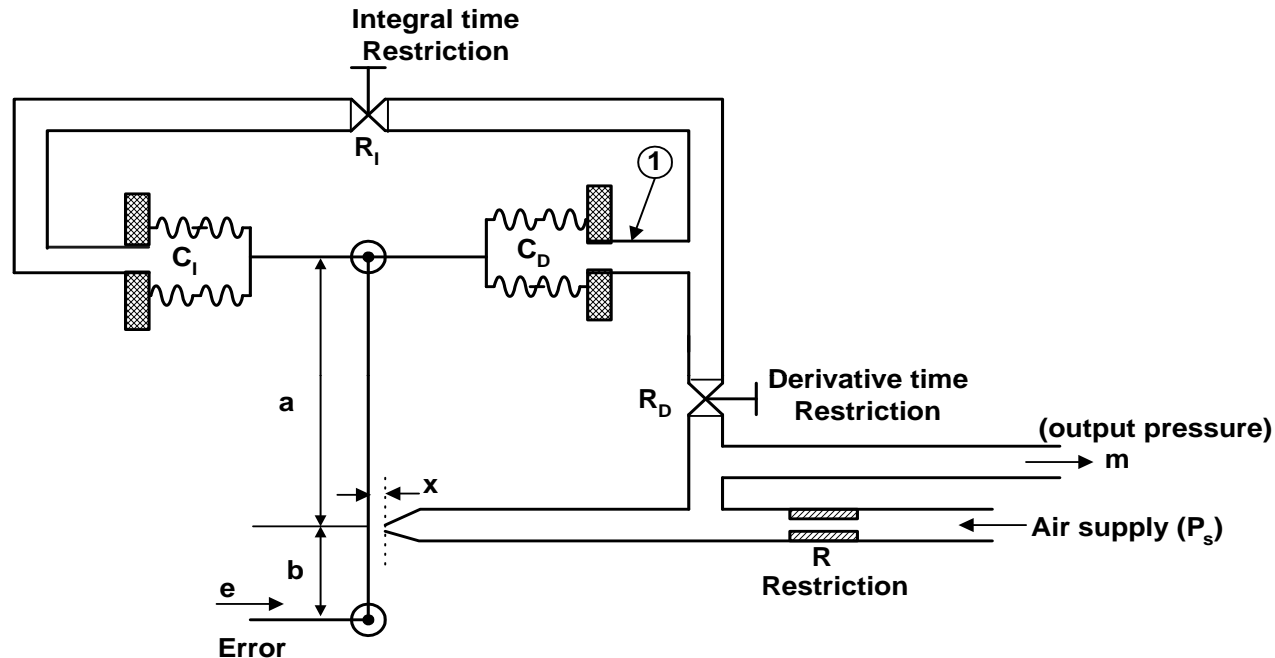
Thus, assuming $K_n \gg 1$, i.e. $\frac{1}{K_n} \approx 0$,

$$\frac{M(s)}{E(s)} = \frac{a}{bK_b} \left(1 + \frac{1}{sT_i} \right) = K_p \left(1 + \frac{1}{sT_i} \right), \quad \text{where} \quad K_p = \frac{a}{bK_b}$$

Pneumatic Proportional-Integral-Derivative Controller



Pneumatic Proportional-Integral-Derivative Controller

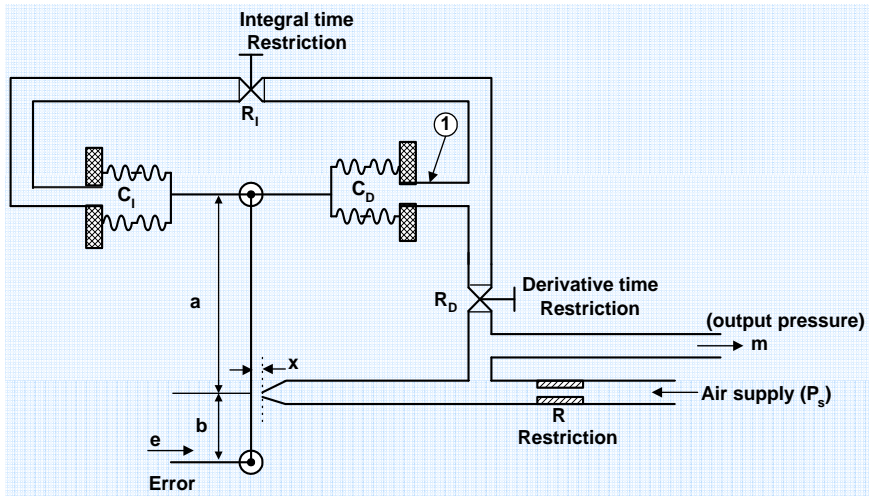


The Baffle-Nozzle separation may be expressed as:

(assuming $R_I \gg R_D \gg R$)

$$\begin{aligned}
 -X(s) &= \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s) \\
 &\quad + \left(\frac{b}{a+b} \right) \left(\frac{K_b}{(1+sT_i)(1+sT_D)} \right) M(s) \\
 &= \left(\frac{a}{a+b} \right) E(s) - \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{(1+sT_D)} - \frac{1}{(1+sT_i)(1+sT_D)} \right) M(s)
 \end{aligned}$$

Pneumatic Proportional-Integral-Derivative Controller

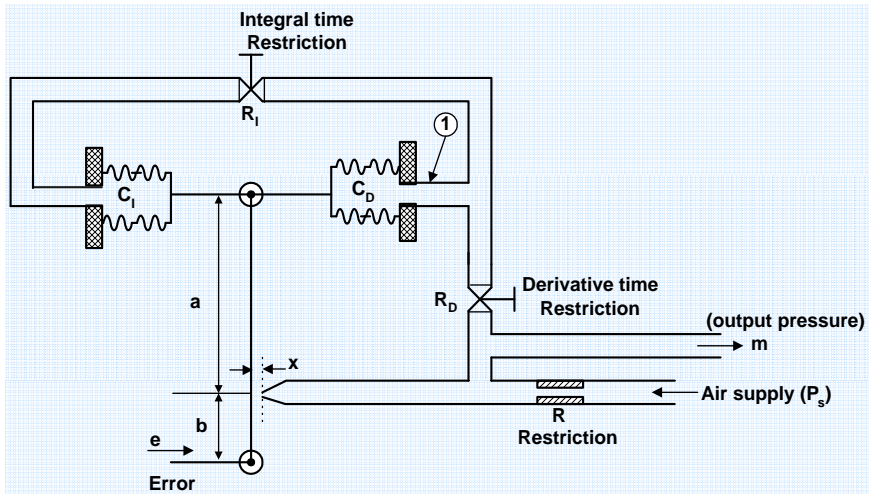


$$\begin{aligned}
 -X(s) &= \left(\frac{a}{a+b}\right)E(s) - \left(\frac{b}{a+b}\right)\left(\frac{K_b}{1+sT_D}\right)M(s) \\
 &+ \left(\frac{b}{a+b}\right)\left(\frac{K_b}{(1+sT_i)(1+sT_D)}\right)M(s) \\
 &= \left(\frac{a}{a+b}\right)E(s) - \left(\frac{bK_b}{a+b}\right)\left(\frac{1}{(1+sT_D)} - \frac{1}{(1+sT_i)(1+sT_D)}\right)M(s)
 \end{aligned}$$

Now,

$$\frac{1}{1+sT_D} - \frac{1}{(1+sT_i)(1+sT_D)} = \frac{1+sT_i-1}{(1+sT_i)(1+sT_D)}$$

Pneumatic Proportional-Integral-Derivative Controller

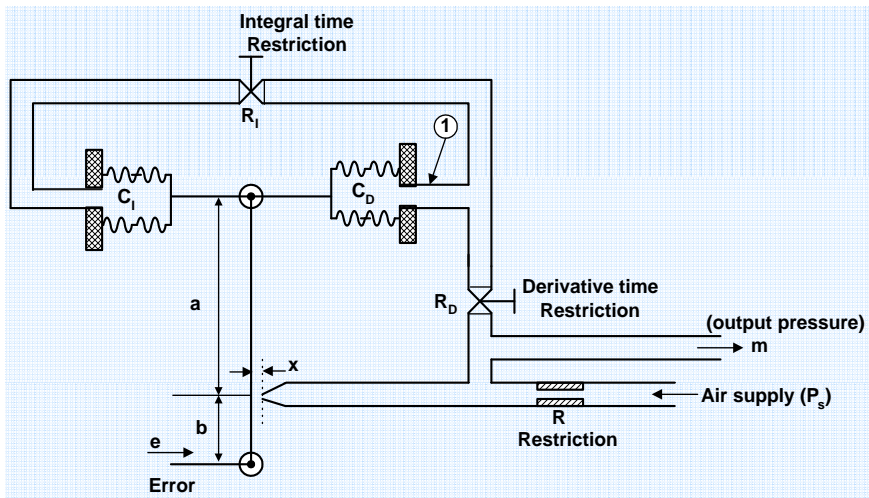


$$\begin{aligned}
 -X(s) &= \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_D} \right) M(s) \\
 &+ \left(\frac{b}{a+b} \right) \left(\frac{K_b}{(1+sT_i)(1+sT_D)} \right) M(s) \\
 &= \left(\frac{a}{a+b} \right) E(s) - \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{(1+sT_D)} - \frac{1}{(1+sT_i)(1+sT_D)} \right) M(s)
 \end{aligned}$$

Now,

$$\begin{aligned}
 \frac{1}{1+sT_D} - \frac{1}{(1+sT_i)(1+sT_D)} &= \frac{1+sT_i-1}{(1+sT_i)(1+sT_D)} \\
 &= \frac{sT_i}{1+sT_i+sT_D+s^2T_iT_D} \\
 &= \frac{1}{1+\frac{1}{sT_i}+\frac{T_D}{T_i}+sT_D}
 \end{aligned}$$

Pneumatic Proportional-Integral-Derivative Controller

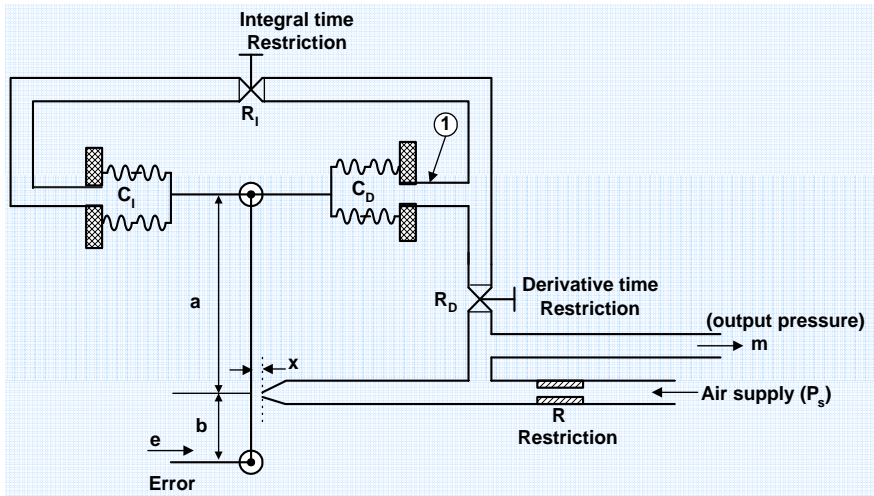


$$\begin{aligned}
 -X(s) &= \left(\frac{a}{a+b}\right)E(s) - \left(\frac{b}{a+b}\right)\left(\frac{K_b}{1+sT_d}\right)M(s) \\
 &\quad + \left(\frac{b}{a+b}\right)\left(\frac{K_b}{(1+sT_i)(1+sT_d)}\right)M(s) \\
 &= \left(\frac{a}{a+b}\right)E(s) - \left(\frac{bK_b}{a+b}\right)\left(\frac{1}{(1+sT_d)} - \frac{1}{(1+sT_i)(1+sT_d)}\right)M(s)
 \end{aligned}$$

Therefore,

$$-X(s) = \left(\frac{a}{a+b}\right)E(s) - \left(\frac{bK_b}{a+b}\right)\left(\frac{1}{1 + \frac{1}{sT_i} + \frac{T_D}{T_i} + sT_D}\right)M(s)$$

Pneumatic Proportional-Integral-Derivative Controller



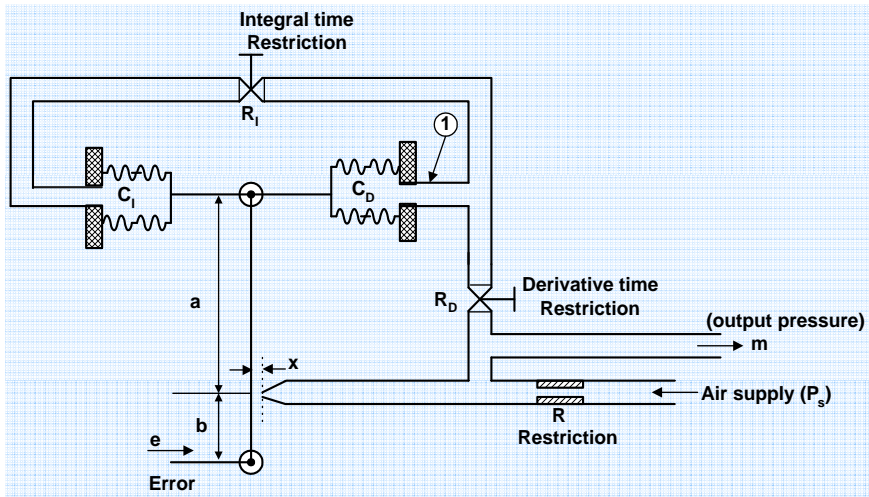
$$\begin{aligned}
 -X(s) &= \left(\frac{a}{a+b} \right) E(s) - \left(\frac{b}{a+b} \right) \left(\frac{K_b}{1+sT_d} \right) M(s) \\
 &\quad + \left(\frac{b}{a+b} \right) \left(\frac{K_b}{(1+sT_i)(1+sT_d)} \right) M(s) \\
 &= \left(\frac{a}{a+b} \right) E(s) - \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{(1+sT_d)} - \frac{1}{(1+sT_i)(1+sT_d)} \right) M(s)
 \end{aligned}$$

Therefore,

$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{1 + \frac{1}{sT_i} + \frac{T_D}{T_i} + sT_D} \right) M(s)$$

Now, $M(s) = -K_n \cdot X(s)$

Pneumatic Proportional-Integral-Derivative Controller



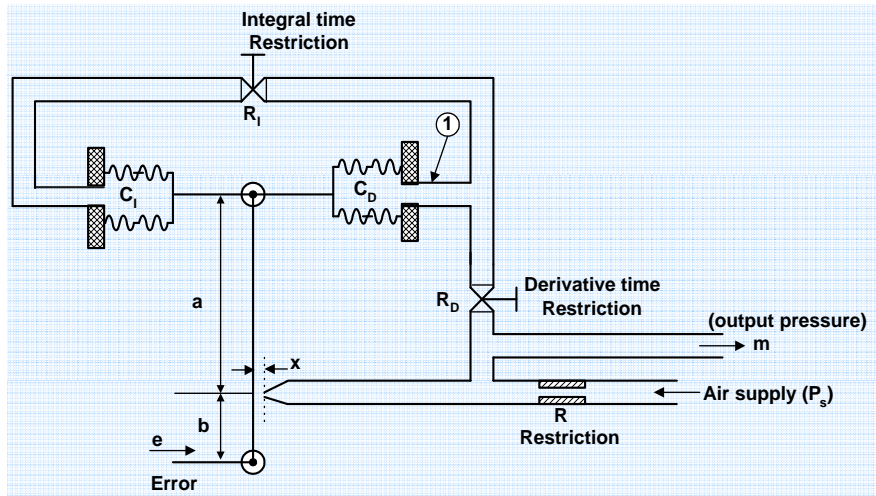
$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{1 + \frac{1}{sT_i} + \frac{T_D}{T_i} + sT_D} \right) M(s)$$

and $M(s) = -K_n \cdot X(s)$

Therefore,

$$\frac{M(s)}{K_n} = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D} \right) M(s)$$

Pneumatic Proportional-Integral-Derivative Controller



$$-X(s) = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{1 + \frac{1}{sT_i} + \frac{T_D}{T_i} + sT_D} \right) M(s)$$

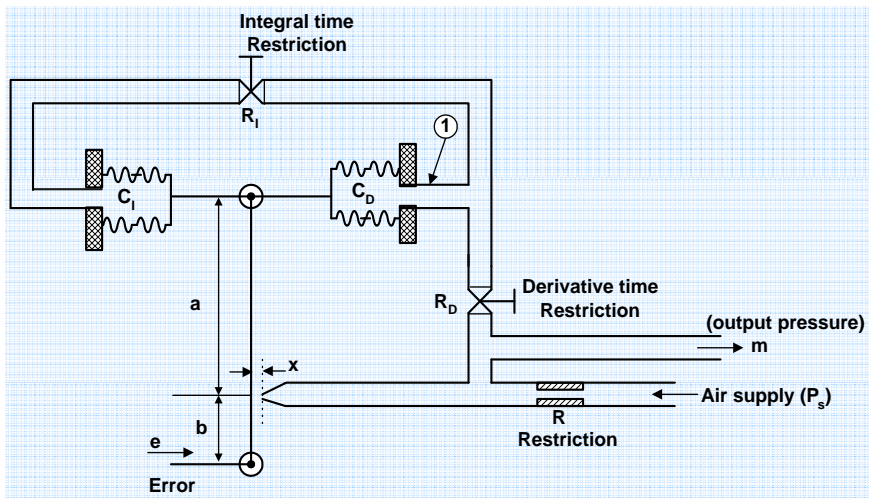
and $M(s) = -K_n \cdot X(s)$

Therefore,

$$\frac{M(s)}{K_n} = \left(\frac{a}{a+b} \right) E(s) - \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D} \right) M(s)$$

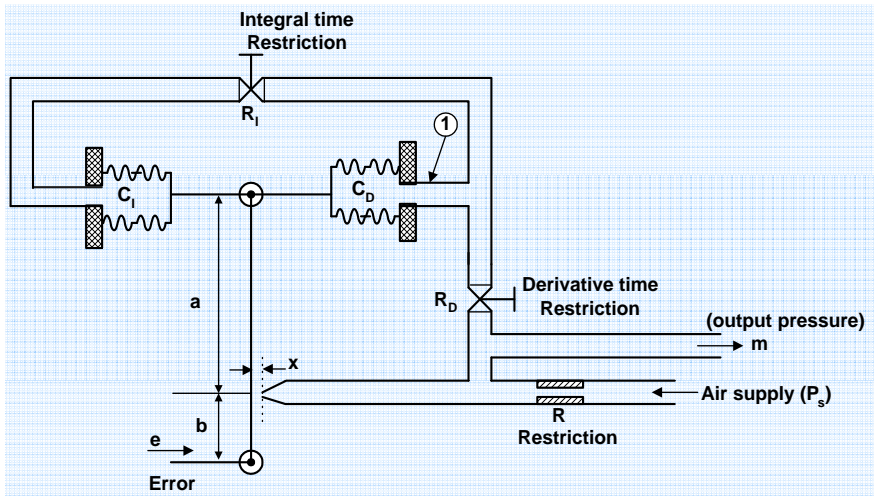
or,
$$M(s) \left[\frac{1}{K_n} + \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D} \right) \right] = \left(\frac{a}{a+b} \right) E(s)$$

Pneumatic Proportional-Integral-Derivative Controller



$$M(s) \left[\frac{1}{K_n} + \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D} \right) \right] = \left(\frac{a}{a+b} \right) E(s)$$

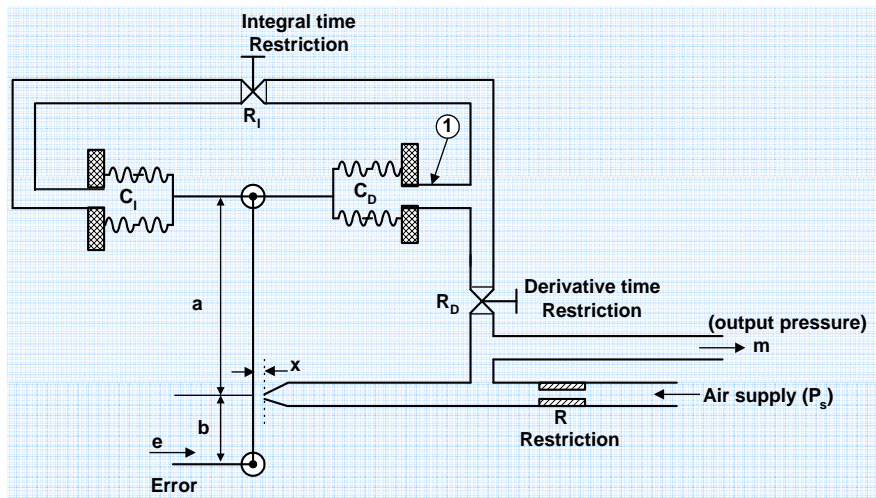
Pneumatic Proportional-Integral-Derivative Controller



$$M(s) \left[\frac{1}{K_n} + \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D} \right) \right] = \left(\frac{a}{a+b} \right) E(s)$$

as $K_n \gg 1$, $\frac{1}{K_n} \approx 0$

Pneumatic Proportional-Integral-Derivative Controller



$$M(s) \left[\frac{1}{K_n} + \left(\frac{bK_b}{a+b} \right) \left(\frac{1}{1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D} \right) \right] = \left(\frac{a}{a+b} \right) E(s)$$

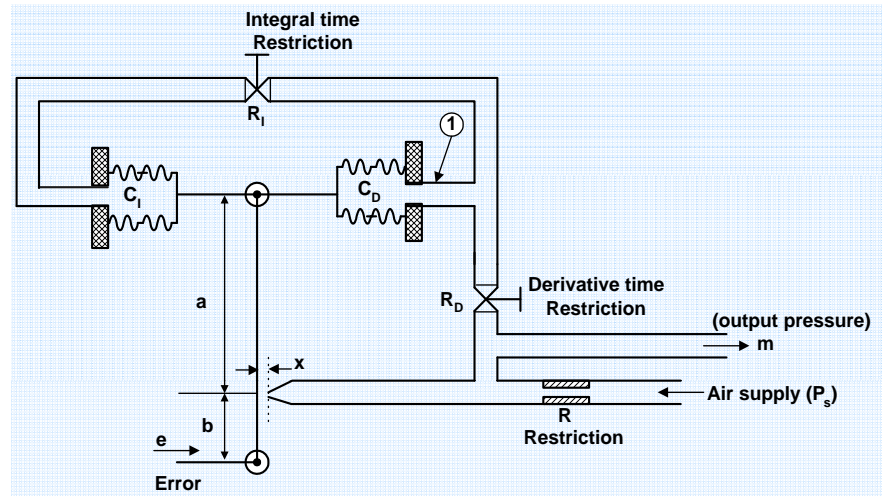
as $K_n \gg 1$, $\frac{1}{K_n} \approx 0$

Therefore,

$$\frac{M(s)}{E(s)} = \left(\frac{a}{bK_b} \right) \left(1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D \right) = K_p \left(1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D \right)$$

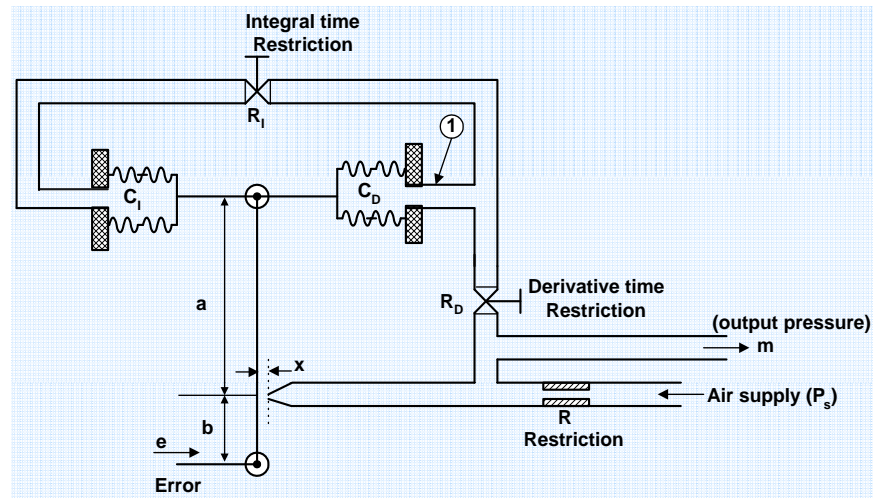
where $K_p = \frac{a}{bK_b}$

Pneumatic Proportional-Integral-Derivative Controller



$$\frac{M(s)}{E(s)} = \left(\frac{a}{bK_b} \right) \left(1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D \right) = K_p \left(1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D \right) \quad \text{where} \quad K_p = \frac{a}{bK_b}$$

Pneumatic Proportional-Integral-Derivative Controller



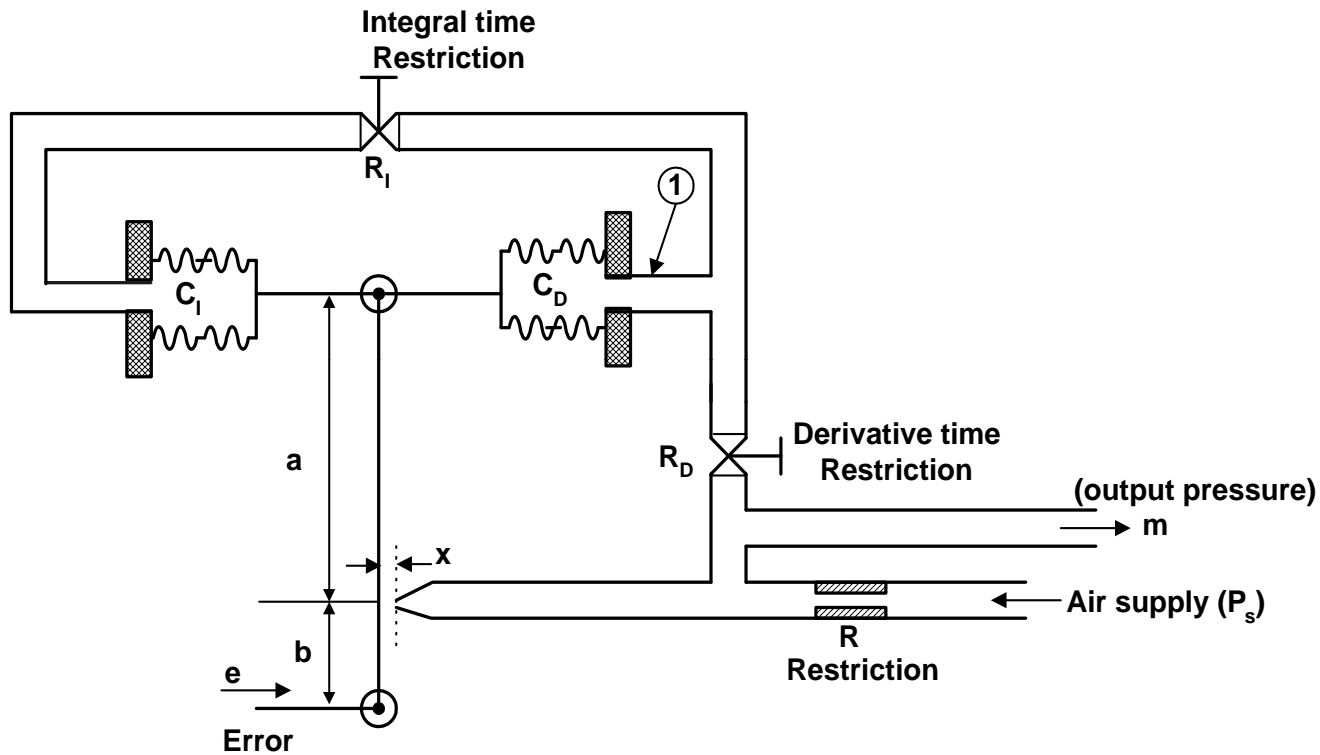
$$\frac{M(s)}{E(s)} = \left(\frac{a}{bK_b} \right) \left(1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D \right) = K_p \left(1 + \frac{T_D}{T_i} + \frac{1}{sT_i} + sT_D \right) \quad \text{where} \quad K_p = \frac{a}{bK_b}$$

Here,

$\left(\frac{T_D}{T_i} \right)$ imposes an interaction between integral and derivative operations of the controller. If we choose $T_i \gg T_D$, the interaction reduces and the transfer function becomes

$$\frac{M(s)}{E(s)} \approx K_p \left(1 + \frac{1}{sT_i} + sT_D \right) \quad \text{the ideal relation of a PID controller.}$$

Pneumatic Proportional-Integral-Derivative Controller



The controller gain becomes infinite when $T_i = T_D$, if derivative time restriction is placed in position (1).

References

1. **Process Control Systems** *by* Shinskey
2. **Automatic Process Control** *by* Eckman
3. **Principles of Process Control** *by* Patranabis
4. **Process Control** *by* Harriott
5. **Process Systems Analysis and Control** *by* Coughanowr and Koppel
6. **Process Control** *by* Pollard
7. **Chemical Process Control** *by* Stephanopoulos
8. **Modern Control Engineering** *by* Ogata
9. **Applied Process Control** *by* Chidambaram

Thank You